

A REVIEW ON FATIGUE MONITORING OF STRUCTURES

ABSTRACT

Significant research has been conducted in the field of structural health monitoring (SHM). The literature on fatigue calculation, fatigue testing, fatigue modelling and remaining fatigue life is also extensive. However, the number of publications related to the fatigue monitoring process is scarce. Thus, a review was undertaken on this topic. Firstly, this paper reviews existing SHM techniques, addresses their principal classifications and presents the main characteristics of each technique, with a particular emphasis on modal-based methodologies. Secondly, a methodology to perform real-time structural fatigue monitoring is proposed, which can be simultaneously combined with other vibration-based SHM techniques to produce a significant increase in the reliability of monitoring techniques. Furthermore, given that fatigue monitoring requires the calculation of stresses at critical points of the structure, a review of stress measurement and estimation techniques is also presented. Finally, the most common techniques used in fatigue assessment for both the time and frequency domains are described.

Keywords: *SHM, fatigue monitoring, damage, structural analysis, failure analysis*

1 INTRODUCTION

Engineering structures are subject to dynamic loadings which can be random (e.g. wind, waves, etc.) or artificial in nature (Jeary, 1998; Kappos, 2001; Simiu and Yeo, 2019). These dynamic loadings generate internal forces, stresses and strains with variable amplitudes, which can lead to fatigue failure (Bolotin, 1999; Schijve, 2008; Suresh, 1998). Fatigue is a progressive process in which each stress cycle causes an incremental increase in damage.

Fatigue design refers to the calculation of fatigue damage accumulated during the lifetime of structures. Well-established practices in fatigue assessment include the determination of stress time histories, the calculation of the fatigue stress spectrum (cycle counting) and the evaluation of total fatigue damage (Bolotin, 1999; Schijve, 2008; Suresh, 1998). In addition to stress-based methods, approaches based on strain, energy and fracture mechanics (Bjørheim et al., 2022) can be used for fatigue assessment. In stationary random processes (Newland, 2005; Wirsching et al., 1995), fatigue analysis can also be addressed in the frequency domain, in which the fatigue stress spectrum is obtained from the spectral moments of stress power spectral densities (PSDs; Benasciutti, 2012; Bishop, 1999; Bishop and Sherratt, 1990; Dalpiaz et al., 2004; Guennec et al., 2014; Slavič et al., 2020; Zigo et al., 2019).

The stresses needed to perform fatigue assessment can be predicted using finite element dynamic analysis in the time or frequency domains. Dynamic behaviour is defined by mass, damping and stiffness matrices (Beards, 1996; Chopra, 2019; Clough and Penzien, 1993) and simplified fatigue loading models from codes and standards (API, 2000; British Standards Institution, 2005; ISO 13819-2:1995, 1995; NORSOK Standar, 2004) are usually considered in fatigue analysis.

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3 However, actual fatigue performance may differ from design calculations. On the one
4 hand, fatigue loading models do not describe actual loads but are selected to represent
5 similar effects as those created by real loadings. On the other hand, discrepancies in mass
6 and stiffness between the numerical model and the real structure are unavoidable due to
7 the difficulty of modelling supports, joints, interaction with fluids, etc. Moreover, all
8 mechanisms related to the damping of structures are difficult to accurately model.
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10 More accurate fatigue life predictions can be achieved if several parameters in the finite
11 element model are updated with experimental data obtained through modal testing
12 (Friswell and Mottershead, 1995; Marwala et al., 2016) to attain good agreement in terms
13 of eigenvalues and eigenvectors. Fatigue life calculations can also be improved if strain
14 time histories are measured with strain sensors or estimated from structural responses
15 obtained using other sensors, such as accelerometers. If experimental responses are used
16 to estimate fatigue in real time, fatigue damage can be monitored.
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19 Two fatigue analysis approaches are usually considered in codes and standards to predict
20 fatigue performance: safe life design and damage tolerance analysis. Safe life is based on
21 preventing damage initiation, and a regular in-service inspection is not needed (British
22 Standards Institution, 2005; Śledziewski, 2017). Meanwhile, damage tolerance analysis
23 is based on good structural fatigue performance in the presence of a defect or damage,
24 and detailed inspection and maintenance planning is required to detect and correct fatigue
25 damage throughout the structure's design service life (British Standards Institution, 2005;
26 Śledziewski, 2017).
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29 Structural health monitoring (SHM) methods are a set of techniques that can be used to
30 detect, locate and assess the extent of damage in engineering structures. They provide an
31 alternative approach to local non-destructive inspection techniques. In SHM, damage is
32 defined as changes to the material and/or geometric properties of a structural system,
33 including changes to boundary conditions and system connectivity (Boller et al., 2009;
34 Chen and Ni, 2018; Crane, 2017; Farrar and Worden, 2012).
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37 According to Rytter and Kirkegaard (Rytter and Kirkegaard, 1994), four levels of
38 monitoring can be defined: detection, location, assessment and prediction. This paper
39 focuses on detection and localisation SHM techniques.
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41 Experimental responses can be used to simultaneously estimate fatigue and detect and
42 localise damage in a structure. Thus, fatigue monitoring can be combined with other SHM
43 techniques to significantly increase the reliability of monitoring techniques used in
44 structures.
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46 Nevertheless, previous reviews of SHM or fatigue failure calculation techniques have
47 typically presented these topics separately, which indicates the necessity of a study that
48 provides a complete overview of the main contributions of both approaches. Thus, the
49 current paper provides a comprehensive overview of existing SHM techniques (Section
50 2), with a particular emphasis on modal-based methodologies (Section 3), and the most
51 common techniques used in fatigue assessment (Section 4).
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54 Furthermore, conclusions from the review process and the main challenges identified in
55 relation to SHM and fatigue assessment techniques are presented, with the expectation
56 that the combination of both approaches in the future would improve the safety and
57 effectiveness of real-time fatigue monitoring for entire structures.
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2 STRUCTURAL HEALTH MONITORING

Structural health monitoring generally refers to any type of damage detection procedure for civil, aerospace or mechanical engineering structures (Dervilis, 2013). Damage can be defined as changes introduced to a system, either intentionally or unintentionally, that affect its current or future performance (Farrar and Worden, 2012). For this reason, SHM is considered an alternative to current local inspection methods, which are more expensive for large structures.

Several review papers on SHM have been published (Amafabia *et al.*, 2017; Bagavathiappan *et al.*, 2013; Barke and Chiu, 2005; Carden and Fanning, 2004; Chang *et al.*, 2003; Ciang *et al.*, 2008; Doebling *et al.*, 1996; Fan and Qiao, 2011; Goyal and Pabla, 2016; Gunes and Gunes, 2013; Li *et al.*, 2014; Lynch, 2006; Mitra and Gopalakrishnan, 2016; Montalvão *et al.*, 2006; Sohn *et al.*, 2003; Ye *et al.*, 2016), which indicates that a rich body of literature exists on this topic. The first general review of SHM was published in 2004 (Sohn *et al.*, 2003). Later, Gunes and Gunes (Gunes and Gunes, 2013) addressed the main damage assessment methodologies and challenges and gaps in SHM. These challenges include the optimisation of sensor number and location, the identification of features that are sensitive to low levels of damage, the ability to identify changes in these features and the development of statistical methods. Li *et al.* (Li *et al.*, 2014) reviewed SHM innovations and applications for infrastructures and proposed some theories and methods for SHM, such as sensing technology, sensor placement, signal processing and data fusion, system identification and damage detection. Review papers on specific applications of SHM have also been published. For example, Barke and Chiu (2005), Montalvão *et al.* (2006), Ciang *et al.* (2008) and Chang *et al.* (2003) addressed SHM applications in relation to the railway industry, composite materials, wind turbines and civil infrastructure, respectively. In addition, Bagavathiappan *et al.* (Bagavathiappan *et al.*, 2013) discussed advances in infrared thermography, Mitra and Gopalakrishnan (Mitra and Gopalakrishnan, 2016) reviewed wave-based SHM and Lynch (Lynch, 2006) focused on sensors for SHM. Reviews on vibration-based SHM have been also undertaken (Carden and Fanning, 2004; Fan and Qiao, 2011).

With regard to SHM techniques, this paper mainly focuses on previous developments in modal-based methods (Section 3) due to their popularity, applicability and robustness in SHM. Nevertheless, before examining modal-based methods, it is useful to review the possible classifications of SHM techniques and explain how information is organised in the paper.

Firstly, SHM techniques can be divided into continuous or intermittent techniques based on the frequency of their application (Cawley, 2018). Intermittent techniques measure responses for specific periods of time; no information is gathered on structural responses the rest of the time. The advantage of this methodology is that the instrumentation can be utilised to monitor other structures. Moreover, experimental responses can be later processed at the office and compared with the undamaged state. By contrast, continuous monitoring requires a more complicated infrastructure since experimental responses must be transmitted in real time to the site where the measurements are processed (also in real time) (Cawley, 2018).

Secondly, SHM methods can also be classified into local and global methods based on the scope of the variables. Local methods use high-frequency ultrasonic waves whose wavelengths should be smaller than the size (Fritzen, 2005). By contrast, global methods typically use the lower modes of the structure.

Finally, SHM methods can be divided into static methods, which are used to measure changes in static responses, and dynamic-based methods, which make use of a structure's vibration properties. This was the classification selected to organise the literature review conducted for this paper (see Fig. 1). In the following subsections, each category of SHM methods is briefly introduced.

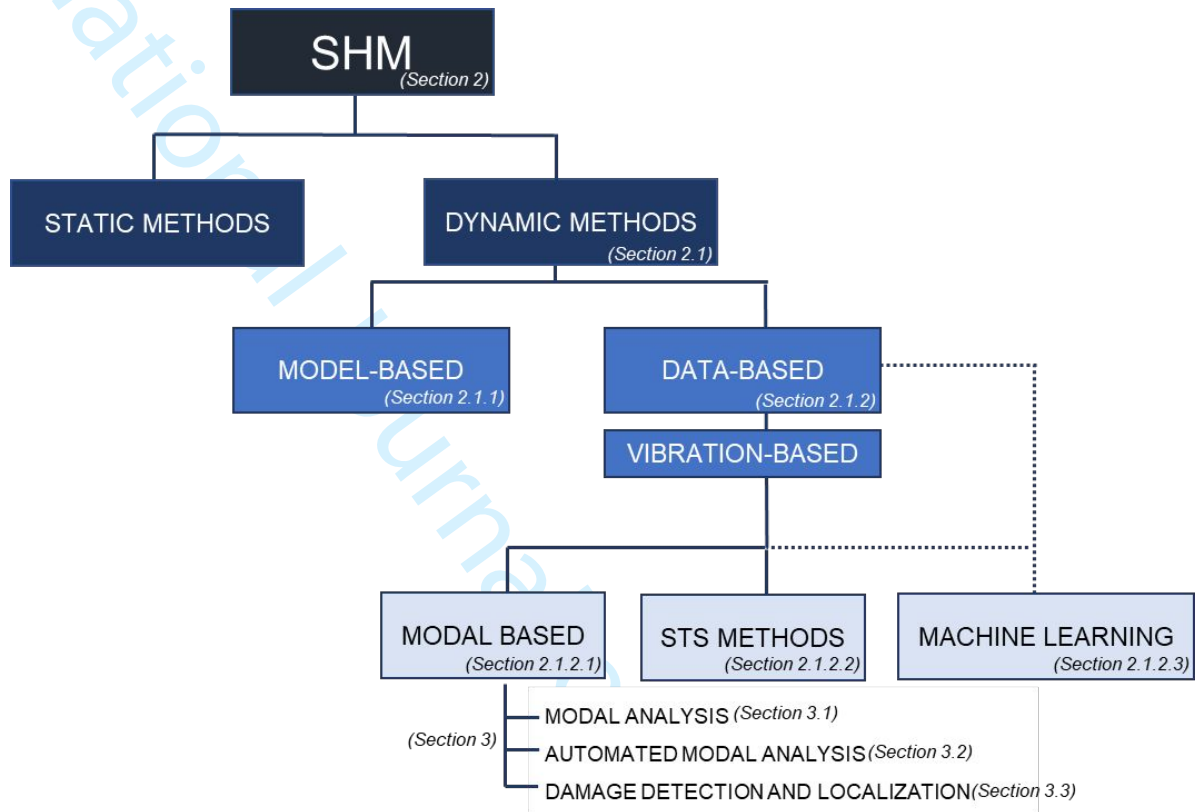


Figure 1. Classification of structural health monitoring (SHM) methods.

2.1 STATIC SHM METHODS

As previously established, static SHM methods are based on the evaluation of changes in static response (i.e. slowly varying parameters over a long period). Although static methods can be used for a wide range of applications, it has been proven that they are a powerful tool in masonry heritage structures. Examples of the most common tests reviewed by Pallarés et al. (Pallarés et al., 2021) include infrared thermography, X-ray imaging, tomography, ultrasound/sonic test, sonic tomography, georadar, acoustic emission, thermography, flat jack tests, endoscopy/videoboroscopy inspection, impact echo testing, coring, hardness tests, penetration tests and ground-penetrating radar. Applications in real structures have also been published, for example, Saisi et al. (Saisi et al., 2016) used static SHM to assess the structural condition of a historic belltower.

Static and dynamic methods have been reviewed by Kralovec and Schagerl (Kralovec and Schagerl, 2020), and the theoretical capabilities of combining such dynamic methods with static methods were also discussed in that paper. Additionally, data processing and analysis is an important subject in static methods, for this reason, Baraccani et al.

(Baraccani et al., 2017) worked on the interpretation of data from static and dynamic SHM.

As previously indicated, static methods lie beyond the scope of this review and are therefore not discussed in detail.

2.2 DYNAMIC SHM METHODS

Dynamic methods use vibration responses to gather information about changes in a structure's dynamic properties, which enables its health to be monitored. Dynamic SHM methods can be classified as model-based SHM or data-based SHM

Model-based techniques enable the construction of a well-correlated model of the structure, which is used to predict the dynamic response of the structure and allows damage to be detected and located. Meanwhile, data-based techniques are used to assess the health of a structure based on the evolution of real data obtained through experimental measurements over time.

In the following subsection, more detailed descriptions of both subdivisions (model-based and data based) and previous research contributions are presented.

2.2.1 Model-based SHM

Model-based SHM consists of constructing a finite element model, which is later used to identify and localize damage in either mass or stiffness (Avendaño-Valencia and Fassois, 2017; Gardner, 2018; Lee and Cho, 2016; Moore et al., 2012; Park and Reich, Gregory W., 1999). However, the accuracy provided by the finite element model depends on the level of correlation with the real structure. Thus, the numerical model must be firstly correlated with test data obtained from the real structure. Then, different techniques, known as model updating, must be applied in order improve the accuracy of the numerical model, which consists of adjusting some parameters of the numerical model in order to minimize the discrepancies (in terms of eigenvalues and eigenvectors) between the numerical model and the real structure, by using test data (see Fig. 2).

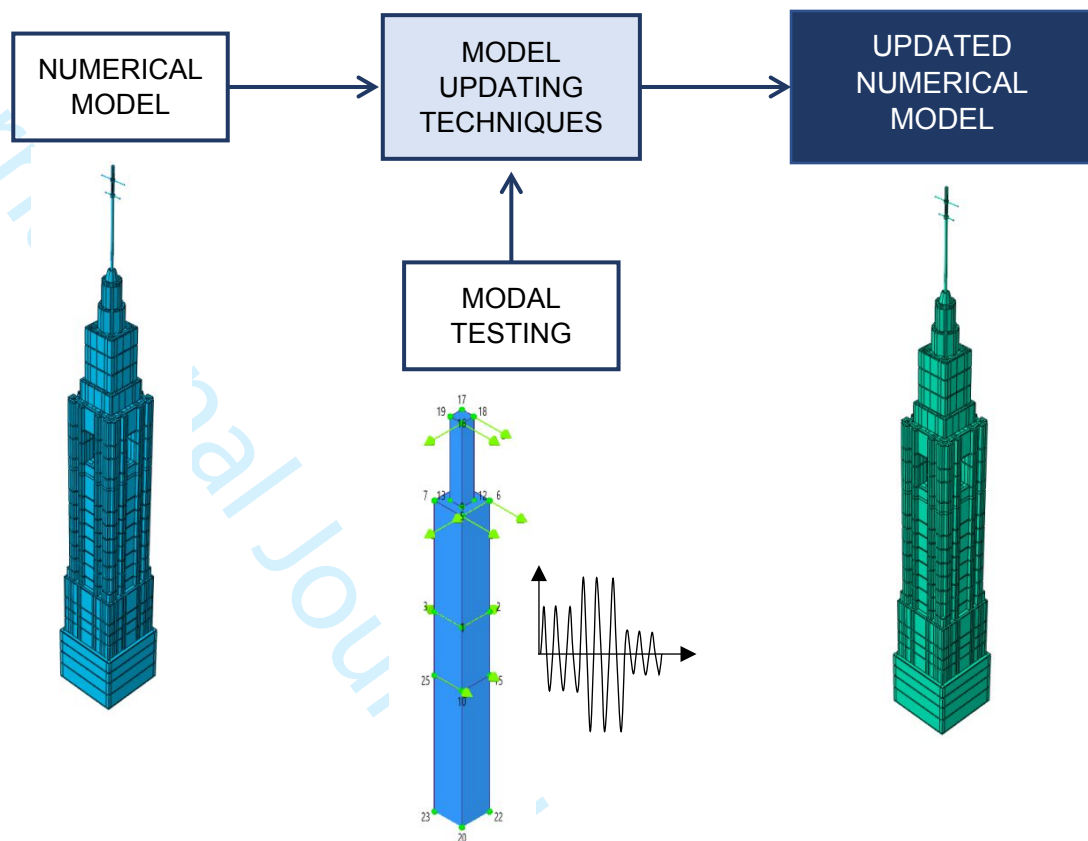


Figure 2. Modal updating process.

Many practical applications of model-based techniques have been published. For example, Stull et al. (Stull et al., 2011) described their use in naval applications, particularly on ship hulls. In addition, Tewolde et al. (Tewolde et al., 2017) applied model-based SHM in relation to wind turbines.

Moreover, there is a vast body of literature on model-based SHM applications in damage detection and localisation (Abdel Wahab and De Roeck, 1999; Ahmadian et al., 1996; Hajela and Soeiro, 1990; Natke and Cempel, 1991; Wahab et al., 1999; Zimmerman and Kaouk, 1994), based on updating certain parameters of the numerical model using experimental modal parameters. Wahab et al. and Zimmerman and Kaouk (Wahab et al., 1999; Zimmerman and Kaouk, 1994) proposed an algorithm based on a minimum rank update theory to provide insights on the location and extent of structural damage. Moreover, Hajela and Soeiro (Hajela and Soeiro, 1990) updated numerical models using static and modal analysis techniques. Finally, Ahmadian et al. (Ahmadian et al., 1996) presented two damage location indicators based on the observation that a change in a particular substructure results in a change in its modes.

2.2.2 **Data-based SHM**

Data-based SHM techniques use real data about a structure obtained through experimental measurements. Data-based SHM involves the observation of a system over time using experimental responses measured through an array of sensors and the extraction and analysis of damage-sensitive parameters (e.g. experimental modal parameters). The structure's undamaged state, which corresponds to a healthy structure, is used as a pattern. Then, data obtained from posterior measurements are compared with the healthy state.

Data-based techniques which rely on the measurement of vibration signals are known as vibration-based methods, and a vast body of literature exists on the topic (Brownjohn et al., 2011; Carden and Fanning, 2004; Deraemaeker et al., 2008; Fan and Qiao, 2011; Fritzen, 2005; Goyal and Pabla, 2016; Khodabandehlou et al., 2019; Magalhães et al., 2012; Ubertini et al., 2016). Goyal and Pabla (2016) reviewed several vibration monitoring and signal processing methods used in SHM. Moreover, Fritzen (2014, 2005) discussed the use of modal information and the direct use of forced and ambient vibrations; they proposed different SHM strategies, depending on the type of measurement data. Brownjohn et al. (2011) reviewed the vibration-based monitoring of civil infrastructure (covering a range of applications, mainly bridges) and highlighted both challenges and successes. Convolution neural networks (Khodabandehlou et al., 2019) and the effect of changing environmental conditions (Deraemaeker et al., 2008) had also been studied. Vibration-based SHM applications have also been reported in the scientific literature in relation to bridges (Brownjohn et al., 2011; Dederichs et al., 2023), arch bridges (Magalhães et al., 2012), historic belltowers (Ubertini et al., 2016) and other structures. The use of actuators, sensing devices and smart sensors in the structure (with on-board computational and communication capabilities) leads to modern concepts of smart structural health monitoring (Fritzen, 2005, 2014). Moreover, in vibration-based SHM, the structural vibration response can be used to detect changes that may indicate damage or degradation (Fritzen, 2014).

In vibration-based methods, temperature, wind velocity, wind direction, wave height, wave direction, humidity and operational conditions (loading conditions, mass loading and speed loading) are known to influence the modal parameters of structures that are subject to these environmental conditions (Chang et al., 2003; Christensen, 2020). The undesired effects of environmental or operational variations must be removed through data normalisation procedures to separate signal changes caused by operational and environmental variations in the system from structural changes of interest, such as structural deterioration or degradation (Kullaa, 2010; Sohn, 2007).

Vibration-based SHM methods can be classified into two main types: modal-based methods and statistical time series (STS) methods (see Fig. 1).

2.2.2.1 **Modal-based methods**

Modal-based methods use one or a set of the following modal parameters: natural frequencies, mode shapes, strain mode shapes and other variables that are dependent on modal parameters (frequency response functions, change in flexibility, etc.).

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3 Modal analysis is a technique used to obtain the modal parameters through experiments
4 (Avitable, 2017; Brincker and Ventura, 2015; Ewins, 2000a; Fu and He, 2001; Heylen *et*
5 *al.*, 2007; Mendes Maia and Montalvão Silva, 1997). Experimental modal analysis
6 (EMA) (Ewins, 2000a; Heylen *et al.*, 2007; Mendes Maia and Montalvão Silva, 1997) is
7 based on input-output system identification and has been used for decades. Modal
8 parameters are estimated from frequency response functions or impulse response
9 functions. Operational modal analysis (OMA) (Au, 2017; Brincker and Ventura, 2015b;
10 Rainieri and Fabbrocino, 2014) is an output-only technique (i.e. only structural responses
11 are used in the estimation process). OMA is attractive in many situations because it does
12 not require excitation to be measured, which is very practical for large structures.
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15 In modal-based SHM, automated modal analysis and automated damage detection
16 techniques must be used because the estimation of modal parameters and the detection of
17 damage must be performed in real time (Andersen *et al.*, 2007; Bajrić *et al.*, 2018;
18 Brincker *et al.*, 2007; Cabboi *et al.*, 2017; Chhipwadia *et al.*, 2000; Rainieri *et al.*, 2007;
19 Rainieri and Fabbrocino, 2010, 2015; Reynders *et al.*, 2012; Sun *et al.*, 2017; Ubertini *et*
20 *al.*, 2013; Wu *et al.*, 2020). Once the damage is detected, techniques to localise and
21 quantify it can be applied.
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26 **2.2.2.2 Statistical time series methods**

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29 STS methods for vibration-based SHM combine random excitation and/or response
30 signals (time series) and statistical and decision-making tools to infer the state of a
31 structure (Fassois and Kopsaftopoulos, 2013). Non-parametric STS methods are based on
32 non-parametric time series representations, such as PSDs, frequency response functions
33 and residual variances. Parametric STS methods are based on time series representations,
34 such as autoregressive moving average models (Fassois and Kopsaftopoulos, 2013).
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38 **2.2.2.3 Machine learning for SHM**

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41 Machine learning (ML) for SHM consists of data-driven approaches (usually vibration-
42 based) which have become popular in recent years due to technological advancements in
43 sensors, high-speed internet and cloud computing. ML is a subcategory of artificial
44 intelligence and refers to a set of algorithms that are capable of learning from available
45 response data by automatically extracting hidden patterns in a large group of data to make
46 predictions (Tibaduiza *et al.*, 2018). ML techniques mainly consist of two steps: feature
47 extraction and training (Azimi *et al.*, 2020; Ye *et al.*, 2019). In several methods, the feature
48 extraction process is based on identifying certain modal parameters from the structural
49 system. Then, the trained ML system is utilised to identify the presence and location of
50 structural damage by performing classification (Avci *et al.*, 2021). Deep learning is a
51 machine learning method commonly used in image recognition (Dong and Catbas, 2021).
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3 MODAL-BASED SHM

Among the SHM techniques presented in the previous section, modal-based SHM is perhaps one of the most popular for the monitoring of civil structures due to recent developments in the field of OMA and the availability of several robust and automated OMA algorithms (Rainieri et al., 2019).

Modal-based SHM methods use modal parameters (natural frequencies, mode shapes and damping ratios) estimated from the experimental responses. In other words, a modal analysis technique must be used to extract modal parameters from the experimental responses. Changes observed in modal parameters with respect to a predefined reference condition are used as indicators of the formation, location and severity of structural damage.

3.1 MODAL ANALYSIS

Modal analysis is used to characterise a structure's dynamic behaviour. Modal analysis separates a structure's response into vibration modes which are defined by the following modal parameters: natural frequencies, mode shapes, damping ratios and modal masses. On the one hand, modal analysis is termed theoretical modal analysis when modal parameters are determined using an analytical model or a numerical model. On the other hand, when modal parameters are determined using an experimental approach, modal analysis is known as experimental modal analysis (EMA) or OMA, depending on the type of excitation used in the experiments. Modal testing encompasses the experimental techniques (vibration testing) used to measure experimental responses (utilised in EMA and OMA) and excitation forces (utilised in EMA). The basic assumptions of modal analysis are linearity, time invariance and observability (see Fig. 3).

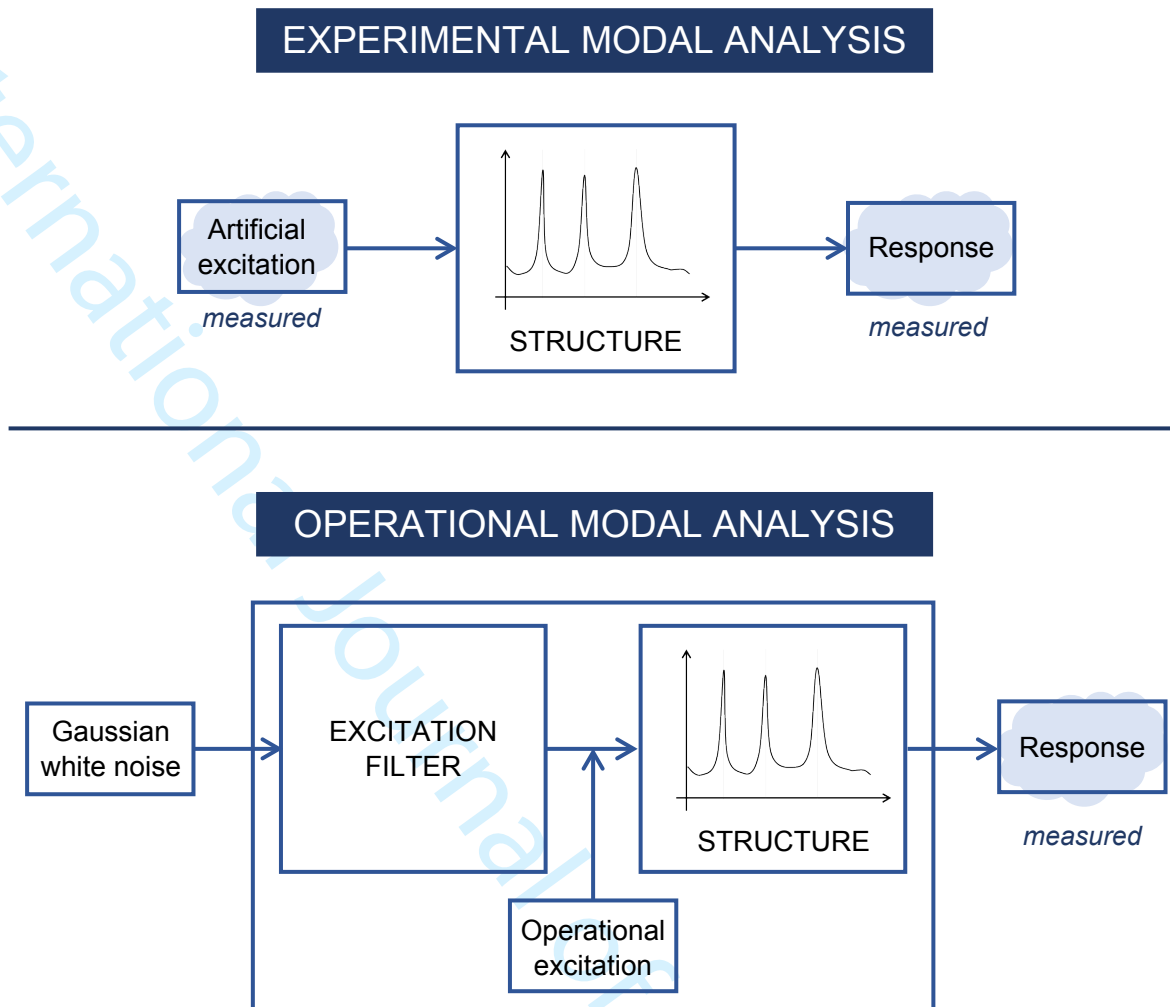


Figure 3. Modal testing techniques: EMA and OMA.

In EMA, both excitation forces and responses must be known to determine modal parameters (Ewins, 2000a; Fu and He, 2001; Heylen et al., 2007; Mendes Maia and Montalvão Silva, 1997). The loading used to excite the structure is commonly artificial, and no other excitation loading is allowed when using this technique.

OMA is used to determine modal parameters without knowledge of input excitation. In short, the forces which are naturally present during the operation of the structure are used as excitation and not measured (Au, 2017; Brincker and Ventura, 2015; Rainieri and Fabbrocino, 2014). A stochastic framework is used in OMA, assuming that the excitation is Gaussian white noise.

When both artificial and operational forces are acting on a structure, OMA and EMA can be combined in the identification process (i.e. both measured and unmeasured forces are considered.) This technique is called operational modal analysis with exogenous input (OMAX; Guillaume et al., 2006) or operational modal analysis with harmonic (OMAH) excitation (Brandt et al., 2019).

Although EMA can be used for SHM, OMA is the most commonly used technique for periodic or continuous monitoring in structures.

3.2 AUTOMATED MODAL ANALYSIS

When using SHM techniques based on modal parameters, periodic or continuous estimation of the latter is needed, which involves a significant amount of user interaction (Neu et al., 2017; Reynders et al., 2012). In SHM, the modal parameter estimation of a single dataset is of little importance because the evolution of modal parameters with time (modal tracking) is the variable of interest. Thus, considerable data must be processed in a short amount of time; thus, methodologies to automatically estimate modal parameters have gained attention in recent years (Andersen et al., 2007; Christensen et al., 2021; Rainieri and Fabbrocino, 2010, 2015; Reynders et al., 2012; Sun et al., 2017). Automated techniques for OMA have also been reported in the literature. For instance, Rainieri and Fabbrocino (2010) developed a new automated algorithm, which was validated in civil engineering structures. Subsequently, Rainieri and Fabbrocino (2015) developed an algorithm for automated estimation of tensile loads in cables and tie rods. Moreover, Reynders et al. (2012) proposed a fully automated, three-stage clustering approach for interpreting stabilisation diagrams. Christensen et al. (2021) compared the poly-reference least squares complex frequency (PLSCF) and the multi-reference Ibrahim time domain (MITD) algorithms and developed a new automated modal analysis algorithm. Finally, Andersen et al. (2007) studied automated modal estimation in large structures, while Sun et al. (2017) examined the topic in the context of a cable-stayed bridge.

Furthermore, several methods have been proposed for automated modal estimation in both the time and frequency domains. They can, in turn, be classified as parametric and non-parametric methods (Avitable, 2017; Brincker and Ventura, 2015; Ewins, 2000a; Heylen *et al.*, 2007; Mendes Maia and Montalvão Silva, 1997). Non-parametric frequency domain methods are based on selecting the peaks of variables (complex mode indicator, function, normalised power spectral density, singular values, etc.) derived from frequency response functions or PSDs (Andersen et al., 2007; Rainieri et al., 2007; Rainieri and Fabbrocino, 2010). Automated parametric methods are based on the automated interpretation of stabilisation diagrams (Christensen and Brandt, 2020; Reynders et al., 2011; Su et al., 2021; Zonno et al., 2017), which involves tracking estimates of modal parameters as a function of model order (Christensen, 2020; Reynders et al., 2012; Ubertini et al., 2013). As the model order is increased, the estimates of physical modal parameters stabilise. Poorly excited modes may not stabilise until a very high model order, whereas very active modes stabilise at a very low model order (Christensen, 2020; Neu et al., 2017; Reynders et al., 2012). Reynders et al. (2012) presented a classification of methods for the automated interpretation of stabilisation diagrams, while Christensen (2020) studied the effect of model order, number of time lag values and starting time lag value in the estimation of modal parameters. Moreover, Zini et al. (2022) proposed a statistical method to automatically define the cut-off thresholds in the hierarchical clustering phase. He et al. (2022) proposed a fully automated unified modal identification, with a focus on structures with many or very few sensors deployed. Dederichs et al. (2023) compared numerous automation algorithms to identify structural modes from the stabilisation diagram using experimental data from the monitored Hardanger Bridge.

Finally, Li et al. (2022) proposed an automated OMA algorithm based on a combination of parametric and non-parametric algorithms: second-order blind identification (SOBI) and covariance-driven stochastic subspace identification (SSI-COV).

3.3 DAMAGE DETECTION AND LOCALISATION

In this section, damage detection and localisation techniques are reviewed in depth.

3.3.1 DAMAGE DETECTION

The most common modal-based techniques used to detect damage are the eigenfrequency method, which is used to monitor changes in natural frequencies, and eigenvector-based criteria, which are used to monitor changes in mode shapes (Frigui *et al.*, 2018) (see Fig.4).

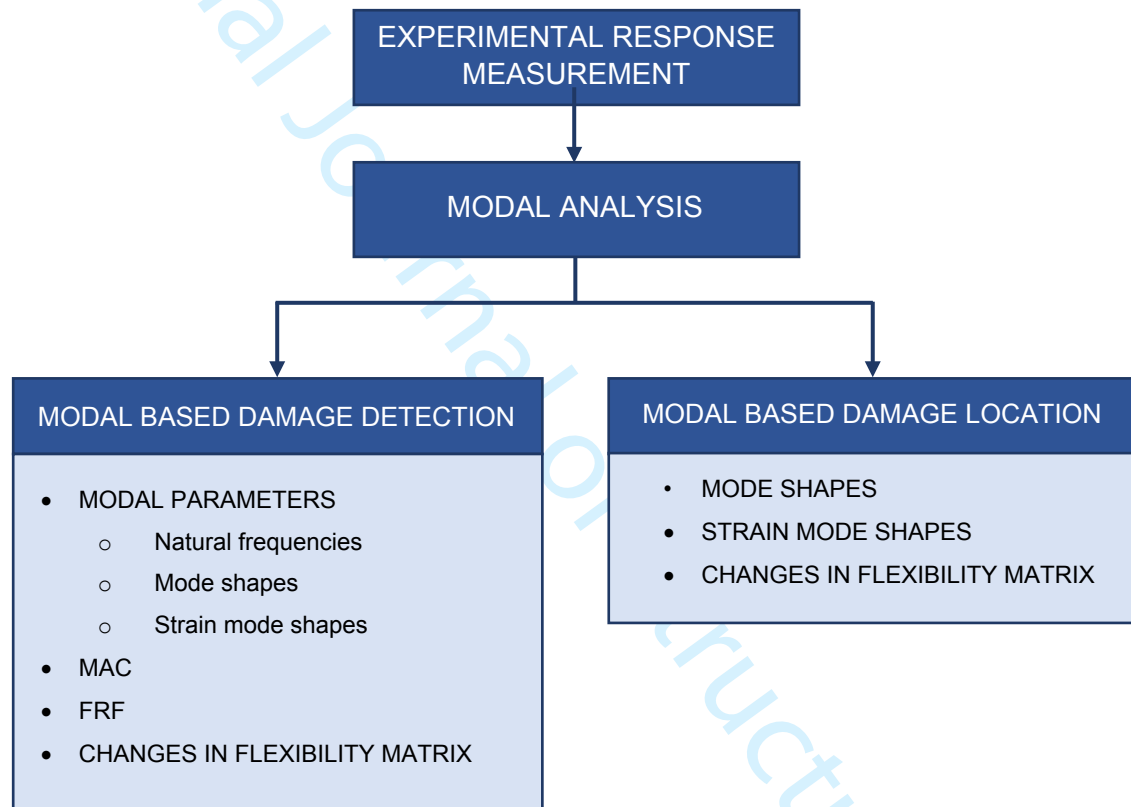


Figure 4. Modal-based damage detection and location methods.

The monitoring of natural frequencies (see Fig. 4) is a simple method and consists of comparing a set of natural frequencies for two states:

$$\Delta f_j = f_{dj} - f_{uj} \quad (1)$$

where j denotes the j -th mode, d the damaged state and u the undamaged state. When using this technique, mode pairing is mandatory.

One of the advantages of natural frequencies is that they are very sensitive to damage. However, they are also sensitive to other mechanical and environmental effects (Frigui et al., 2018). It is well-known that changes in temperature, wind velocity, wave height, wind direction and wave directionality modify natural frequencies (Christensen, 2020; Li et al., 2009; Peeters and De Roeck, 2001; Ubertini et al., 2017). Moreover, boundary conditions, which depend on the soil type, can affect modal parameters.

The criteria based on eigenvectors compares a set of mode shapes (see Fig. 4) (Ewins, 2000b; Lein and Beitelschmidt, 2014). A mode pairing is also mandatory. The best-known method is the modal assurance criterion (MAC) (Allemang, 2003; Allemang and Brown, 1982; Fotsch and Ewins, 2000; Rigner et al., 1998; Vacher et al., 2010), which compares the shapes of two eigenvectors based on the inner product of both vectors. If two vectors ϕ_{di} (damage state) and ϕ_{uj} (undamaged state) are compared, MAC (ϕ_{di}, ϕ_{uj}) is given by the following equation:

$$MAC(\phi_{ui}, \phi_{dj}) = \frac{|\phi_{ui}^T \phi_{dj}|^2}{(\phi_{ui}^T \phi_{uj})(\phi_{di}^T \phi_{dj})} \quad (2)$$

The first paper about MAC was published in 1982 by Allemang and Brown (1982). Allemang (Allemang, 2003) reviewed the historical development of the original MAC, along with other related assurance criteria proposed between 1982 and 2002. Moreover, typical abuses of MAC were also identified. MAC can also be applied in the frequency domain (Fotsch and Ewins, 2000) and extended to complex modes (Vacher et al., 2010). Lein and Beitelschmidt (2014) conducted a comparative study of different model correlation methods.

Mode shapes are affected by damages, however, for low severity damages the method indicates damage only in higher-order modes, which are more sensitive to damage but also more difficult to identify in real-life situations. Moreover, the estimation of mode shapes is not as precise as the estimation of natural frequencies (Frigui et al., 2018). West (1984) used MAC to detect structural changes. Fox (1992) compared the use of natural frequencies and mode shape data to detect damage.

Techniques based on monitor strain mode shapes have also been proposed in the literature (see Fig. 4). They are based on the relationship between mode shape curvatures and flexural stiffness (i.e. when a structure is damaged, its stiffness decreases and induces a variation in mode shape curvature) (Frigui et al., 2018). The modal curvatures of the lower modes are generally more accurate than those of higher modes (Frigui et al., 2018). Pandey et al. (1991) used changes in strain mode shapes to detect damage. In addition, Abdel Wahab and De Roeck (1999) detected damage using modal curvatures and applied this technique to a real bridge.

Changes in frequency response functions or flexibility can be also used to detect damage. The change flexibility matrix can be computed using experimental modal parameters as follows:

$$[\Delta f] = \sum_{r=1}^N \frac{\phi_{ru} \phi_{ru}^T}{\omega_{ru}^2} - \sum_{r=1}^N \frac{\phi_{rd} \phi_{rd}^T}{\omega_{rd}^2} \quad (3)$$

where ϕ_r denotes the r -th mode shape, ω_r the r -th natural frequency, d the damaged state, u the undamaged state and N the number of modes.

The computation of flexibility matrices from vibration data requires mass-normalised mode shapes. If OMA is used to estimate modal parameters, an additional technique to scale the mode shapes is needed (Aenlle and Brincker, 2019, 2014; Aenlle et al., 2007a, 2007b; López-Aenlle et al., 2005, 2010). López-Aenlle et al. (2005) proposed some methods to determine scaled mode shapes. Modal scaling was performed using a finite element model in a study by Aenlle and Brincker (2014) and using a mass change strategy in studies by Aenlle et al. (2007a, 2007b) and López-Aenlle et al. (2010). Furthermore, Aenlle et al. (2007a) proposed a methodology to optimise the number, location and magnitude of attached masses, which was validated for a cantilever beam by Aenlle et al. (2007b). Damage detection through changes in modal flexibility has been studied by Pandey and Biswas (1994, 1995) and Toksoy and Aktan (1994). Finally, Salawu (1995) applied the integrity index method to a concrete highway bridge.

3.3.2 DAMAGE LOCALISATION

Modal-based damage localisation methods (see Fig. 4) are traditionally based on changes in mode shapes, mode shape derivatives or the flexibility matrices assembled from available modes (Doebling et al., 1996; Fox, 1992; Pandey et al., 1991; Pandey and Biswas, 1994, 1995; Salawu, 1995; Stubbs and Kim, 1996; Toksoy and Aktan, 1994; West, 1984).

Mode shapes can be easily estimated using modal analysis, but the localisation of damage based on the curvature of mode shapes has been shown to be more sensitive to damage than mode shapes (Abdel Wahab and De Roeck, 1999; Pandey et al., 1991).

Shokrani et al. (2018) used mode shape curvatures for damage localisation under varying environmental conditions. Frigui et al. (2018) proposed a new parameter called curvature damage factor (CDF), in which the difference in modal curvature between the damaged and undamaged states is averaged over all modes. If the structure contains several damaged locations, CDF provides a clear identification of these locations. When the method is applied to real structures, irregularities in measured mode shapes can arise, and a curve fitting must be applied before calculating modal curvature.

Bernal (Bernal, 2006; Bernal and Gunes, 2002, 2004) proposed a damage localisation method based on changes in the experimental flexibility matrix. The technique identifies the damaged elements of a structure as belonging to the set of elements whose internal forces under the action of a certain set of load vectors (designated as damage location vectors or DLVs) are zero. These vectors define a basis for the null space of the change in flexibility.

4 FATIGUE MONITORING

In this section we deal with the steps needed to perform continuous fatigue monitoring in structures. The organization of this part of the paper is summarized in Figure 5. Firstly, we introduce the most common techniques used to predict fatigue failure. Then the available techniques to estimate stresses are discussed. Finally, the methods for fatigue assessment in both time and frequency domains are reviewed and analysed.

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2
3 Well-established practices in fatigue assessment include the determination of stress time
4 histories, the calculation of the fatigue stress spectrum (cycle counting) and the evaluation
5 of total fatigue damage (Bolotin, 1999; Schijve, 2008; Suresh, 1998). In the time domain,
6 the fatigue stress spectrum is obtained from time stress histories using counting
7 algorithms such as the rainflow (Johannesson, 2002; Lindgren and Rydén, 2002; Rychlik,
8 1996a, 1996b) or reservoir methods (BS5400: Part 10, 1980). Fatigue damage is
9 estimated using a damage accumulation model, the best-known being the Palmgren-
10 Miner model (Miner rule; Miner, 1945). However, different damage accumulation
11 models have been proposed in the literature (Benkabouche et al., 2015; Fatemi and Yang,
12 1998).

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14
15 Stress-based models are mainly used to predict fatigue life for high-cycle fatigue, whereas
16 strain-based models are suitable for low-cycle fatigue in which plastic deformation is
17 significant. Energy-based models can consider out-of-phase hardening behaviour because
18 both the stress and strain terms are inherent in the energy expression (Wei and Liu, 2020).
19 Moreover, for welded details, a fatigue approach based on nominal or geometrical stress
20 is preferred to local approaches based on continuum mechanics (Leonetti et al., 2021).

21
22
23 A combination of cycle counting and the Miner rule is generally accepted as one of the
24 best time domain methods for fatigue life estimation. However, this methodology
25 presents some drawbacks in random loadings because the stress time history is only
26 known in a statistical manner, which can be overcome by simulating time histories from
27 the random process. In addition, another negative aspect to consider is the inability to
28 handle load sequence effects. Different studies (Branco et al., 2022; Fiedler and
29 Vormwald, 2016) have examined sequence effects and the consequences of neglecting
30 them.

31
32
33 Stress-based fatigue assessment may be applied using nominal stresses, hot spot stresses
34 or local stresses (Bolotin, 1999; Schijve, 2008; Suresh, 1998). Nominal S-N curves are
35 considered with fatigue life based on nominal and local stresses, whereas hot-spot S-N
36 curves should be used with hot-spot stresses. Several investigations of the hot-spot stress
37 approach have been conducted in recent years. For example, Viana et al. (2019)
38 performed a fatigue assessment based on hot-spot stresses obtained from global dynamic
39 analysis and local static sub-modelling using finite element models. Moreover, Bao et al.
40 (2022) proposed an indirect method for evaluating hot-spot stresses induced by complex
41 load conditions.

42
43
44 In stationary random processes (Newland, 2005; Wirsching et al., 1995), fatigue analysis
45 can also be addressed in the frequency domain; the fatigue stress spectrum is obtained
46 from the moments of stress PSDs (Benasciutti, 2012; Bishop, 1999; Bishop and Sherratt,
47 1990; Dalpiaz et al., 2004; Guennec et al., 2014; Slavič et al., 2020; Zigo et al., 2019).
48 This technique is much more rapid than a transient dynamic analysis in the time domain.
49 Frequency fatigue can be used in combination with FEM software to evaluate fatigue
50 after the loading is known and the dynamic analysis has been performed. As many authors
51 consider the rainflow method to be the most accurate, frequency domain cycle counting
52 techniques attempt to obtain a cycle distribution according to the rainflow counting
53 method in the time domain (Benasciutti, 2012; Slavič et al., 2020).

54
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56 Continuous fatigue monitoring means calculating accumulated fatigue damage in real
57 time during the period that the structure is in operation (Česnik et al., 2012; Jiang et al.,
58 2015; Mršnik et al., 2013). Fatigue monitoring consists of three steps:
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- Localisation of the structure's hot spots to identify the most probable locations of fatigue damage, defined during the fatigue design analysis.
- Measurement or estimation of stresses at the hot spots. Stresses can be measured using appropriate sensors, which are continuously monitored, or estimated from structural responses measured (in real time) using displacement, velocity or acceleration sensors. Stress estimation is explained in detail in the following sections.
- Calculation of accumulated fatigue damage and remaining fatigue life.

Fatigue monitoring can be combined with other SHM techniques. For instance, experimental responses can be used to simultaneously estimate stresses and to detect and localise damage in the structure using a vibration-based SHM technique, which significantly increases the reliability of monitoring techniques used in the structure (i.e. by providing redundancy).

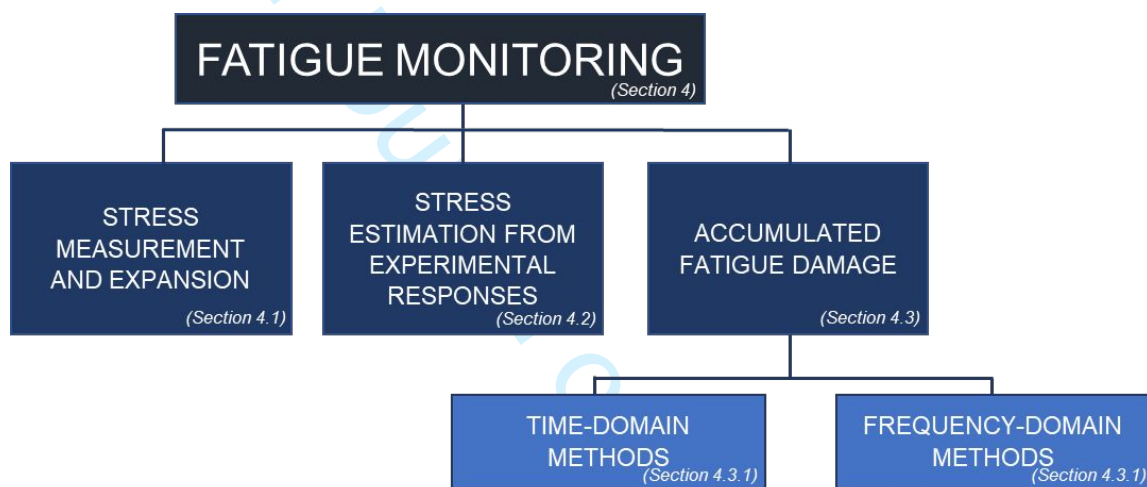


Figure 5. Organization of the fatigue monitoring section.

4.1 STRESS MEASUREMENT AND EXPANSION

Most fatigue analysis techniques consider stresses (stress range, stress amplitude and maximum stress) as the principal variable responsible for fatigue damage. However, no sensors that measure stresses can be found in the market. Therefore, stresses are indirectly obtained from strain measurements. The most common devices for measuring strain are strain gages, such as Wheatstone Bridge Circuit (four-element strain gauge bridge circuit), fibre Bragg grating sensors (Kong et al., 2022; Rao, 1997) and integrated circuit piezoelectric (ICP) strain sensors (Fujimoto et al., 2003). Fibre Bragg grating sensors have several advantages; they are small, lightweight, do not require electrical connections and are compatible with non-invasive remote sensing (Sahota et al., 2020). However, temperature compensation for strain error through thermal fluctuation is essential. On the other hand, ICP sensors have superior signal-to-noise ratio and high frequency noise rejection compared to conventional strain gages (Sirohi and Chopra, 2000).

Using strain sensors, accumulated fatigue damage can be calculated at the stress measurement points by using appropriate stress concentration factors. In addition, strain mode shapes can be obtained through strain modal analysis if an appropriate distribution of sensors is considered, which can be expanded to unmeasured degrees of freedom (DOFs) to estimate stresses at any point of the structure. Moreover, strain measurements can be effective for crack detection because of the sensitivity of strain change to the opening and closing of a crack. However, sensors may not provide adequate information for fatigue crack monitoring since their small size hinders their ability to cover an adequate surface for areas that are prone to fatigue cracks (Taher et al., 2022).

The steps needed to perform real-time stress estimation using experimental strain measurements are shown in Figure 6. It is assumed that the strain mode shapes are known from experiments or a numerical model (1). Firstly, the hot spots (2) to be monitored and the sensor location (3) strategy must be defined. The measured strains $\varepsilon_{xa}(t)$ (4) are then used to estimate the strain modal coordinates $q_\varepsilon(t)$ (5) using the experimental strain mode shapes ϕ_{ε_a} (1). The strains $\varepsilon(t)$ (6) at the selected points of the structure are estimated with the expanded strains mode shapes and the modal coordinates $q_\varepsilon(t)$. Finally, stress time histories $\sigma(t)$ (7) are calculated using $\varepsilon(t)$ and the stress-strain relationship.

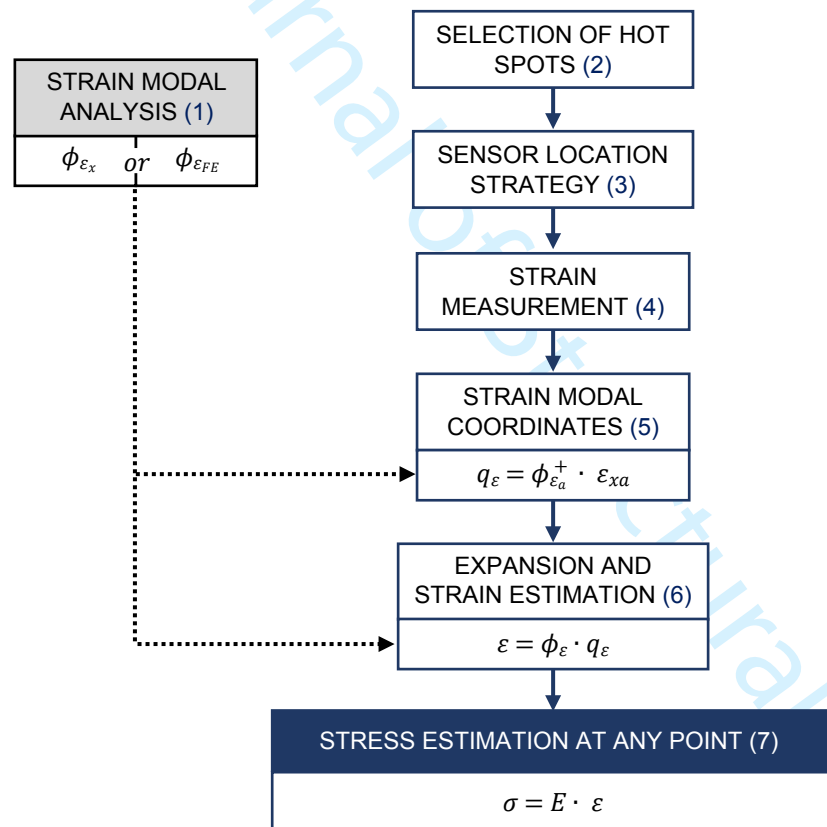


Figure 6. Real-time stress estimation using strain measurements.

4.2 STRESS ESTIMATION FROM EXPERIMENTAL (NON-STRESS) RESPONSES

Stress estimation from experimental (non-stress) responses are techniques commonly based on modal superposition, which allows the modal coordinates to contain information about natural frequencies and damping. Numerical or experimental mode shapes can be used. As experimental mode shapes are only known at a reduced number of points, they are expanded to the unmeasured DOFs using a finite element model. Thus, a numerical model is needed to estimate stresses, but only a good numerical-experimental correlation in terms of mode shapes is required. Different types of finite element models can be used; only a good dynamic correlation is needed, mainly in terms of mode shapes (Fernández et al., 2009; Pelayo et al., 2015).

Brincker et al. (Brincker et al., 2003) published a paper on several potential applications of OMA, such as SHM, load estimation and vibration-level estimation. Moreover, to reduce uncertainty from loading modelling in fatigue calculations, a methodology to estimate stresses at any point of the structure using its experimental response was presented (Hjelm et al., 2005; Pelayo et al., 2015).

The methodology allows stresses to be estimated at any point of the structure using experimental displacement, the structure's velocity or acceleration responses, which are measured at several discrete points (Aenlle et al., 2013; López-Aenlle et al., 2013).

The structure's experimental response can be measured in displacement, velocity or acceleration formats. Accelerometers can be classified into three main groups: piezoelectric, capacitive or piezoresistive accelerometers. Piezoelectric accelerometers have very low noise and offer superior performance to the capacitive and piezoresistive. Capacitive MEMS (micro-electro-mechanical systems) accelerometers are cheap; small; and exhibit high vibration, shock and temperature resistance. However, they have poor temperature characteristics. Finally, piezoresistive accelerometers require amplifiers and temperature compensation, but they have a very wide bandwidth and low noise characteristics (Hanly, 2016).

When measuring displacements, laser Doppler vibrometers are widely recognised as valid measurement tools for structural dynamic measurements (Warren, Niezrecki, et al., 2011). Displacement measurements using the FFT-DDI (Direct Digital Integration) method with different types of accelerometers have been studied (Ribeiro et al., 2003). The performance of laser Doppler vibrometers, digital image correlation and accelerometers have been investigated by Rossi et al. (2002; Warren, Niezrecki et al. (2011; and Warren, Pingle et al. (2011).

The theory needed to estimate stresses has been published in several papers (Aenlle et al., 2013; Fernández et al., 2009; Hjelm et al., 2005; López-Aenlle et al., 2013; Pelayo et al., 2015). If the structure's response is measured with accelerometers, the acceleration modal coordinates $\ddot{q}_x(t)$ can be obtained as follows:

$$\ddot{\mathbf{u}}_x(t) = \mathbf{\Phi}_x \cdot \ddot{\mathbf{q}}_x(t) = \sum_{r=1}^{N \text{ modes}} \boldsymbol{\phi}_{xr} \cdot \ddot{q}_{xr}(t) \quad (4)$$

where subindex x indicates experimental parameters, $\ddot{\mathbf{u}}_x(\mathbf{t})$ is the measured acceleration vector and Φ_x is the experimental mode shape matrix, which can be estimated using EMA or OMA.

Normal stresses in a beam at coordinate x can be estimated with the following equation (Aenlle et al., 2013; Fernández et al., 2009; Hjelm et al., 2005; López-Aenlle et al., 2013; Pelayo et al., 2015):

$$\sigma(\mathbf{x}, \mathbf{t}) = \mathbf{y} \cdot \sum_{r=1}^{N \text{ modes}} \phi_{\epsilon xr}(\mathbf{x}) \cdot q_{xr}(\mathbf{t}) \quad (5)$$

where q_{xr} is the r -th displacement modal coordinate obtained through integration, y is the distance to neutral beam axis and $\phi_{\epsilon xr}(\mathbf{x})$ is the strain mode shape of the r -th mode, which is related (in beams), with mode shapes ϕ_x as follows:

$$\phi_{\epsilon x} = \frac{d^2 \phi_x}{dx^2} \quad (6)$$

The experimental mode shapes must be expanded to unmeasured DOFs to estimate the strain mode shapes at the desired points. A finite element model can be used to this end; experimental mode shapes are expressed as a linear combination of FE mode shapes (Brincker et al., 2014).

The steps to perform real-time stress estimation using the structure's displacement, velocity or acceleration response is shown in Figure 7. It is assumed that the strain mode shapes are known from experiments or a numerical model (1). Firstly, the hot spots (2) to be monitored and the sensor location (3) strategy must be defined. The measured responses (displacements, velocities or accelerations) (4) are then used to estimate the strain modal coordinates $q(t)$ (5). The strains $\epsilon(t)$ (6) at the selected points are estimated with the expanded strain mode shapes and the modal coordinates $q(t)$. Finally, stress time histories $\sigma(t)$ (7) are calculated using $\epsilon(t)$ and the stress-strain relationship.

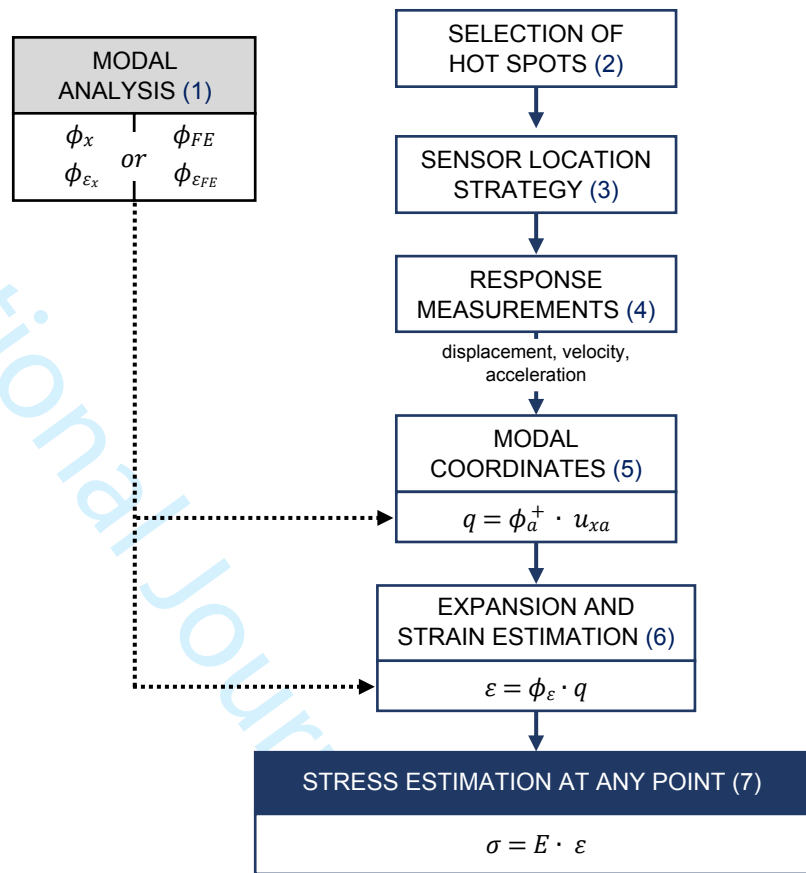


Figure 7. Real-time stress estimation using experimental (non-strain) responses.

This technique has been validated through experimental tests on lab-scaled structures (Fernández *et al.*, 2009; Hjelm *et al.*, 2005; López-Aenlle *et al.*, 2013; Tarpø *et al.*, 2020) and real structures (Christensen, 2020; Dascotte *et al.*, 2013; Henkel *et al.*, 2019, 2020; Hjelm *et al.*, 2005; Iliopoulos *et al.*, 2017; Nabuco *et al.*, 2020; Noppe *et al.*, 2018).

Papadimitriou (Papadimitriou *et al.*, 2011) proposed a methodology for estimating damage resulting from fatigue using spectral methods and operational vibration measurements at a limited number of locations. Using the experimental response time history and a model of the structure, a Kalman filter approach was used to predict the PSDs of stresses in the entire body of the structure. This technique was validated using analytical and numerical models and by considering simulated measurements.

In addition, Gulgec *et al.* (Gulgec *et al.*, 2020) proposed a technique to estimate strain responses using experimental acceleration responses as inputs for a multistage deep neural network based on long short-term memory and fully connected layers. This technique was validated in a lab-scaled structure.

4.3 ACCUMULATED FATIGUE DAMAGE

Once stresses are estimated for identified hot spots where failure is most probable, the corresponding accumulated fatigue damage must be estimated. Using data from constant amplitude tests to estimate fatigue damage under variable amplitude loading requires a cycle-counting method, a damage accumulation law and consideration of the load

sequence effect (Kondo, 2003). To this end, material fatigue characterisation based on the S-N field – or, equivalently, the ε -N field – allows fatigue lifetime to be analytically defined as a previous step in damage assessment. Alternatively, fracture mechanics can also be used to estimate crack growth until its critical value is reached.

Methodologies applied in fatigue damage assessment were traditionally formulated in both the time and frequency domains, which is explained in detail in the following subsections and summarised in Figure 8.

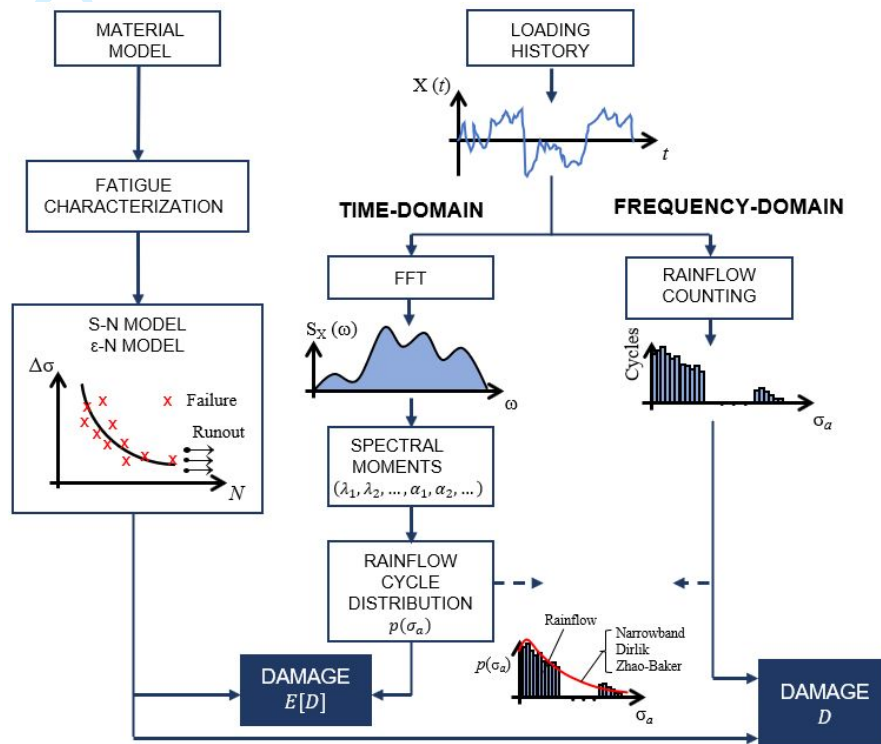


Figure 8. General flowchart of fatigue damage assessment using both time and frequency domain approaches.

4.3.1 TIME DOMAIN METHODS

Time domain methods were formulated first and applied more frequently than frequency domain methods. The analysis of loading history is accomplished using different counting algorithms, such as the rainflow method developed by Matsuishi and Endo (Matsuishi and Endo, 1968), to obtain an equivalent set of counted cycles with constant amplitudes. This allows fatigue damage to be estimated. Despite being widely used as a reference procedure, the rainflow method entails some important disadvantages, such as its dependence on a particular time window selected in the loading history and time-consuming nature.

An extensive literature review on the proposed damage variables and methodologies can be found in the literature (Castillo et al., 2007; Fatemi and Yang, 1998; Santecchia et al., 2016). Five different categories can be distinguished in these proposals:

- *Linear damage models.* Palmgren (Palmgren, 1924) first proposed the linear damage rule and Miner (Miner, 1945) subsequently popularised it as one of the

most widely applied approaches to calculating damage due to its easy formulation, which is only based on the ratio between the applied cycles n_i and the total cycles to failure N_i for the i -th load level:

$$D = \sum_{i=1}^k \frac{n_i}{N_i} \quad (7)$$

where D is a damage index ($0 \leq D \leq 1$) and k is the number of different stress levels.

Linear accumulation models use the S-N curve from constant amplitude tests; they assume no load sequence effects and no damage for stress repetitions below the fatigue limit (Kondo, 2003).

Manson and Halford (Manson and Halford, 1986) proposed the double linear damage rule. Double linear models were also proposed by Langer (Langer, 1937) and Grover (Grover, 1960), separating the initiation and propagation stages. Despite being widely applied, its main drawbacks are independence with respect to both load level and load sequence.

In variable amplitude loading, stress repetitions below the fatigue limit also cause damage. Haibach's rule (Haibach, 1970) and the Corten–Dolan rule (Corten and Dolan, 1956) have been proposed to account for this effect.

- *Non-linear damage models.* In an attempt to improve the incongruences of linear damage rules, Richart and Newmark (Richart and Newmark, 1948) and Marko and Starkey (Marco and Starkey, 1954) proposed the first non-linear damage rule based on a powering cycle ratio to x_i variable for the i -th loading:

$$D = \sum_{i=1}^k \left(\frac{n_i}{N_i} \right)^{x_i} \quad (8)$$

A non-linear modification to Miner's rule for damage accumulation was proposed to reduce the scatter between experimental fatigue life and fatigue life predicted using the Miner rule (Blacha, 2021).

Si-Jian et al. (2018) proposed a non-linear fatigue damage accumulation model which considered the effects of loading history and loading sequence under multi-level stress loading using the S-N field and the Miner rule.

- *Energy-based damage models.* As an alternative to previous phenomenological approaches, different authors have proposed energy-based definitions of fatigue damage, the Smith-Watson-Topper parameter (Watson et al., 1970) being one of the most widely recognised:

$$D = \frac{4\sigma'_f}{E} (2N_N)^{2b_1} + 4\sigma'_f \varepsilon'_f (2N_N)^{b_1 + c_1} \quad (9)$$

where σ'_f and ε'_f are fatigue strength and ductility coefficients, respectively; N_N is the number of reversals to failure; and b_1 and c_1 are constants that depend on an instantaneous strain-hardening law.

- Continuum-based damage models. This novel approach addresses the continuum mechanical behaviour of a medium in degenerating conditions, which was originally stated by Kachanov (Kachanov, 1984) and Rabotnov (Rabotnov, 1969). Thanks to Chaboche and Lesne (Chaboche and Lesne, 1988), this proposal has been popularised as a highly non-linear damage rule that takes into account the mean stress effect:

$$D = 1 - \left[1 - \left(\frac{n}{N_N} \right)^{1/1 - \alpha} \right]^{1/\beta - 1} \quad (10)$$

where α is a function of the stress state and β is a material function.

- Probabilistic damage models. Finally, probabilistic approaches have recently appeared due to the work of Fernández-Canteli (Castillo et al., 2007; Fernández-Canteli et al., 2014). In it, the classical Miner's rule was converted to a random variable from which the statistical distribution of the number of cycles to failure and the stress range can be numerically computed (Castillo and Fernández-Canteli, 2009):

$$p = 1 - \exp \left[- \left(\frac{(\log N - B)(\log \Delta \sigma - C) - \lambda}{\delta} \right)^\beta \right] \quad (11)$$

where B and C are the horizontal and vertical asymptotes (that is, the cycle value below which failure does not occur and the fatigue endurance limit, respectively), while λ, β and δ are the location, shape and scale Weibull parameters, respectively.

4.3.2 FREQUENCY DOMAIN METHODS

Frequency domain or spectral methods (Bishop, 1999; Gao and Moan, 2008; Quigley et al., 2016; Sherratt et al., 2005; Zalaznik and Nagode, 2011) allow complex loading histories to be directly and rapidly computed as part of a more consistent statistical and analytical approach than time domain methods.

Loading history is classified as a random narrow-band (NB) process or a broad-band (BB) process (Wirsching et al., 1995). The former leads to simpler and easier formulations about statistical properties, while the latter offers more complex identification of stress cycles. In this sense, the statistical information contained in the spectral density $S_X(\omega)$ of a random process X can be summarised by means of the m -th spectral moments λ_m as follows:

$$\lambda_m = \int_{-\infty}^{\infty} \omega^m S_X(\omega) d\omega \quad m = 0, 1, 2, \dots \quad (12)$$

where the even moments are directly related to the variance σ_X^2 of the random process and its derivatives, as in $\lambda_0 = \sigma_X^2$. From a statistical perspective, the rainflow cycle distribution could be considered a bivariate distribution with maximum and minimum stresses, $p_{RFC}(\sigma_{max}, \sigma_{min})$, or, equivalently, with mean and amplitude stresses, $p_{RFC}(\sigma_a, \sigma_m)$ (Benasciutti and Tovo, 2005, 2006). Indeed, one of the most relevant differences between frequency and time domain approaches is that the former use analytical definitions for the rainflow cycle distribution in fatigue damage assessment.

However, due to the inherent complexity of pairing procedures for peak-to-valley in the rainflow algorithm, there is no explicit analytical solution for the bivariate rainflow cycle distribution (Benasciutti and Tovo, 2006; Lalanne, 2013). Thus, the bivariate distribution is usually simplified by neglecting the mean stress effect and considering only the stress amplitude; as a result, different approximate proposals in the literature are defined in terms of $p_{RFC}(\sigma_a)$ instead.

Amongst these proposals, three are most widely applied:

- *Narrow-band approximation* is based on the assumption that the random process is of NB type; that is, each peak and valley is coincident with each cycle. Thus, the stress amplitude can be considered to follow a Rayleigh distribution:

$$p_{RFC}^{NB}(\sigma_a) = \frac{\sigma_a}{\sigma_X^2} \exp\left[-\frac{1}{2}\left(\frac{\sigma_a}{\sigma_X}\right)^2\right] \quad (13)$$

- *The Dirlik model* (Dirlik, 1985; Dirlik and Benasciutti, 2021) suggests a mixture distribution between an exponential component and two Rayleigh components:

$$p_{RFC}^{DK}(\sigma_a) = \frac{1}{\sigma_X} \left[\frac{D_1}{Q} \exp\left(-\frac{Z}{Q}\right) + \frac{D_2 Z}{R^2} \exp\left(-\frac{Z^2}{2R^2}\right) \right] \quad (14)$$

where $Z = \sigma_a/\sigma_X$ is the normalised amplitude and D_1, D_2, Q and R are constants that depend on the spectral moments.

- *The Zhao and Baker model* (Zhao and Baker, 1992) proposes a mixture distribution with Rayleigh and Weibull components:

$$p_{RFC}^{ZB}(\sigma_a) = w\alpha\beta Z^{\beta-1} \exp(-\alpha Z^\beta) + (1-w)Z \exp\left(-\frac{Z^2}{2}\right) \quad (15)$$

where w is a weighting factor ($0 \leq w \leq 1$) as a function of the spectral parameters and α and β are the scale and shape Weibull parameters, respectively.

- *Tovo-Benasciutti* (Dirlik and Benasciutti, 2021; Tovo, 2002) proposed that the amplitude–mean joint probability distribution of rainflow cycles lies between two limit distributions and can be estimated as their linear combination:

$$p_{RFC}^{TB}(\sigma_a) = \sigma_a p_{LCC}(\sigma_a, m) + (1 - w) p_{RC}(\sigma_a, m) \quad (16)$$

where w is a weight factor that must be determined. The two functions $p_{LCC}(\sigma_a, m)$ and $p_{RC}(\sigma_a, m)$ represent the amplitude–mean distributions of the level-crossing counting (LCC) and of the simple-range counting (RC).

Once the rainflow cycle distribution for the stress amplitude has been analytically defined, the fatigue damage assessment can be performed. By considering the Basquin law ($s^k N = C$) from the material characterisation step, the expected rainflow damage rate \bar{D}_{RFC} (i.e. damage/sec) can be calculated as follows (Rychlik, 1993):

$$\bar{D}_{RFC} = \nu_a C^{-1} \int_0^{\infty} \sigma_a^k p_{RFC}(\sigma_a) d\sigma_a \quad (16)$$

where ν_a is the rate of occurrence of counted cycles (that is, counted cycles per second) and $p_{RFC}(\sigma_a)$ can be defined according to the previously mentioned proposals. Finally, from Equation (16), total expected damage D until failure can be directly obtained as follows:

$$D = \bar{D}_{RFC} T_f \quad (17)$$

where T_f is the time to failure (that is, the fatigue lifetime). Moreover, it should be noted that, depending on the particular analytical definition of the rainflow cycle distribution in Equation (16), the total expected damage in Equation (17) could be different.

5 DISCUSSION AND CONCLUSIONS

This paper was mainly motivated by the lack of comprehensive literature on the combination of SHM and fatigue assessment techniques. Thus, a general overview of existing SHM techniques was presented, with a particular emphasis on modal-based methodologies and the most common techniques used in fatigue assessment.

Given that the literature on both topics is usually presented separately, the main discussions and conclusions in this paper are divided into three subsections: (a) SHM, (b) fatigue monitoring and (c) a general overview that includes both approaches.

Each subsection summarises the main contributions and tendencies that have emerged in recent decades in relation to each topic and the main challenges for future research identified by the authors.

5.1 Structural Health Monitoring

The most frequently used SHM techniques are modal-based methods, which assume that structural damage can be detected from changes in one or a set of modal parameters. The development of automated modal identification techniques in recent years has contributed to an increase in the use of these techniques.

Despite the numerous benefits of SHM, there are some challenges that this technique must overcome. Damage is in most times a local phenomenon, whereas many SHM

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3 techniques try to detect global damage and, consequently, damage might not be detected.
4 The key challenge in the SHM is to avoid false alerts (false positives or false negatives)
5 which reduce the confidence of the SHM techniques.
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7 On the one hand, an improvement of the existing technologies and methods involved in
8 the monitoring process (type of sensors, location of sensors, sensor resilience, effects of
9 nonlinearities, nonstationary methods for SHM, removal of the environmental effects,
10 data processing, identification techniques, big data, statistical analysis, etc.) will result in
11 a better and more accurate damage prediction with SHM.
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13

14 Cost is another challenge in SHM. Nowadays SHM is expensive in large structures and
15 the owners need to be convinced that the cost will not exceed the benefits given by the
16 SHM.
17

18 Environmental and operational conditions are known to influence the modal parameters
19 of structures. These undesired effects must be removed through data normalisation
20 procedures to eliminate the effect of changes caused by operational or environmental
21 variations. Although some techniques have been developed to remove the effect of
22 temperature, better techniques are needed to remove the effect of other variables such as
23 wind speed, wind direction, wave height, wave direction, etc.
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27 SHM must provide an automated and real-time assessment of a structure; modal-based
28 SHM uses automated modal analysis identification techniques, but other techniques such
29 as model based SHM are more difficult to automate. Thus, another challenge is the
30 application of machine learning techniques for damage identification and localization
31 under unsupervised learning mode.
32
33

34 **5.2 Fatigue Monitoring**

35 Firstly, the literature review on stress measurement techniques for fatigue assessment
36 leads to the conclusion that real-time stress time histories can be estimated from
37 experimental strains measured with strain sensors, which must be expanded to
38 unmeasured DOFs using an expansion method. Sensors based on Bragg gratings are
39 expensive due to the need of a sophisticated interrogation system. The presence of electric
40 and/or magnetic fields can superimpose electrical noise on the strain measurements when
41 using strain gage. On the other hand, different papers in the literature have demonstrated
42 that the use of strain gages or fibre Bragg grating sensors to measure experimental stresses
43 could lead to significant errors if the influence of temperature change is disregarded. For
44 this reason, different methods have been introduced to compensate for the influence of
45 temperature change and reduce the thermal effect on measurements, which can also be
46 mitigated by using self-temperature-compensation strain gages.
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52 Secondly, a literature review on stress estimation techniques which are based on modal
53 superposition methods was presented. These modal superposition methods rely on
54 experimental or numerical mode shapes and the modal coordinates estimated from the
55 structure's experimental response (displacement, velocity, or acceleration). The measured
56 response must also be expanded to unmeasured DOFs. One advantage of this technique
57 is that information about natural frequencies and damping ratios is contained in the modal
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3 coordinates; in other words, this technique is less sensitive to changes in environmental
4 or operational conditions than SHM based on modal parameters.
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6 Finally, regarding the performance of fatigue analysis in the time domain, it can be
7 concluded that the most common practice is to assume a Basquin linear S-N field,
8 combined with the rainflow algorithm to perform the cyclic counting and the Miner Rule
9 to accumulate the damage. Nevertheless, there are more advanced models that include
10 non-linear behaviours, energetic approaches, and probabilistic aspects; based on their
11 promising results, they are likely to become more relevant in the future. Regarding the
12 performance of fatigue analysis in the frequency domain, the Miner rule is also commonly
13 used, but the fatigue stress spectrum is obtained from the moments of the stress PSDs,
14 which provide deeper insights on the problem than the information provided by a rainflow
15 algorithm, which is performed in the time domain. However, most frequency domain
16 techniques proposed in the literature can only be applied to linear systems that are subject
17 to stationary Gaussian processes.
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23 **5.3 General Overview (Both Approaches)**

24 Despite the great advancements in knowledge about fatigue failure that could potentially
25 result from combined works on SHM and continuous fatigue monitoring of real
26 structures, very few studies to date have simultaneously examined both techniques.
27

28 On the one hand, works on SHM that were identified in the literature typically focus on
29 introducing new models to detect or localise failure but do not calculate the accumulation
30 of fatigue damage. When they do, the fatigue models used tend to be the simplest ones,
31 which are based on uniaxial stress amplitude and the Miner rule.
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34 On the other hand, works on the improvement of fatigue assessment are usually centred
35 on presenting new critical parameters (based on stress, strain, energy, etc.), cyclic
36 counting techniques or accumulated damage models (linear, exponential, energetic, etc.).
37 Nevertheless, their application is usually demonstrated through simplified examples with
38 constant amplitude loading (or constant loading steps), and SHM techniques are not used
39 in parallel.
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42 This has led to excellent models in both fields – SHM and fatigue assessment – but
43 separately. For this reason, future research should focus on demonstrating the potential
44 of combined methodologies, which could improve the real-time fatigue monitoring of
45 entire structures and understanding of the causes of fatigue failure.
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