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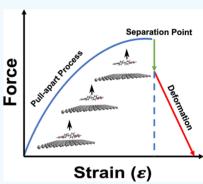
# <sup>1</sup> Mechanical Response of Polymer Epoxy/BMI Composites with <sup>2</sup> Graphene and a Boron Nitride Monolayer from First Principles

<sup>3</sup> Geeta Sachdeva,\* Álvaro Lobato, Ravindra Pandey,\* and Gregory M. Odegard

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4 **ABSTRACT:** Polymer composites possess an integrated combination of structures and 5 properties associated with the host matrix and the fiber material and thus hold the 6 potential of being high-strength materials. In general, the load transfer from the matrix to 7 the fiber depends upon the strength of bonding at the interface, which characterizes the 8 mechanical strength. In this work, first-principles calculations based on the density 9 functional theory are employed to provide the molecular-level description of the interface 10 formed by resins (i.e., diglycidyl ether of bisphenol A (DGEBA) and 4'-11 bismaleimidodiphenylmethane (BMPM)) or hardeners (i.e., diethyl toluene diamine 12 (DETDA) and o,o'-diallyl bisphenol A (DABPA)) with graphene (or boron nitride (BN) 13 monolayer). The results show that the interaction strength between a resin (or hardener) 14 and graphene is mainly governed by the nature of bonding at the interface, and 15 subsequently, the mechanical response follows the hierarchical order of the interface 16 strength at the interface; the transverse stiffness of BMPM/graphene is higher than that of



17 DGEBA/graphene. Moreover, the change in the polarity of the surface from graphene to the BN monolayer improves the superior 18 interfacial strength and thereby a higher transverse stiffness of both resin and hardener composites at the molecular level. These 19 results emphasize the need to use computational modeling to efficiently and accurately determine molecular-level polymer/surface 20 combinations that yield optimal mechanical performance of composite materials. This is especially important in the design and 21 development of high-performance composites with nanoscale reinforcement.

22 KEYWORDS: mechanical properties, epoxy, BMI, graphene, BN monolayer, density functional theory

## 1. INTRODUCTION

23 In a fiber-polymer composite system, fibers play an important 24 role of load-bearing components, and the polymer matrix 25 provides support to fibers by maintaining their orientations in 26 the composite. In recent years, the use of carbon nanotubes 27 (CNT) as a fiber element has helped to make significant 28 advances in transport technology due to their low weight-to-29 stiffness ratio, resistance to environmental impact, and  $_{30}$  processability.<sup>1-3</sup> The C-C covalent bond in carbon rings, 31 which is regarded as the most stable chemical bond in nature, 32 endows graphitic surfaces with excellent mechanical proper-33 ties,<sup>4,5</sup> thereby making them promising candidates for the next 34 generation of high-performance composites. For example, 35 Young's modulus of CNTs with a density of  $1.33 \text{ g/cm}^3$  can be 36 as high as 1 TPa.<sup>6–8</sup> Therefore, focused scientific efforts have 37 been invested in engineering CNT-based composites that 38 exploit the outstanding mechanical properties of individual 39 CNTs. However, the stiffness and strength of individual CNTs 40 have generally not transferred well to large-scale assemblages. To address the abovementioned challenge, both exper-41 42 imental<sup>9-12</sup> and theoretical<sup>13-16</sup> methods have been used to 43 investigate the interface formed by a CNT fiber with the 44 polymer matrix to gain insight into its role in determining the 45 mechanical response of the composite. For example, it has

been shown that sliding occurs within the fractured fibers of 46 long CNTs leading to a poor load transfer,<sup>17</sup> and multiwalled 47 CNTs are often deformed and twisted, which increases their 48 flexibility.<sup>18</sup> One weakness of CNT assemblages is the 49 inadequate noncovalent interaction between adjacent circular 50 CNTs, which leads to the slippage of CNTs during 51 deformation.<sup>17</sup> Hence, it was recently proposed to fabricate 52 self-assembled stacks of large-diameter (~10 nm) CNTs, 53 referred to as flattened CNTs (flCNTs), which maximizes the 54 contact area between CNTs, and thus the noncovalent type of 55 bonding.<sup>19</sup> Therefore, it appears possible that the mechanical 56 properties of the flCNT composite may be further enhanced 57 by carefully engineering its interface.

Also, flattened CNTs, which are referred to as a "dog-bone"- 59 type collapse in the scientific literature, have been linked to 60 improved mechanical performance because of their similarity 61 to graphitic carbon.<sup>20</sup> They can be considered as stacked layers 62

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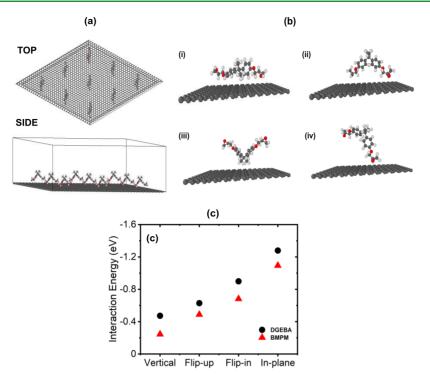


Figure 1. (a) Top and side views of a complex. (b) Molecular orientations: (i) in-plane, (ii) flip-in, (iii) flip-up, and (iv) vertical over graphene. Atomic color codes—O (red), C (gray), and H (white). (c) Orientation-dependent interaction energy of DGEBA (red) and BMPM (black) interacting with graphene.

 $^{63}$  of graphene, which is a two-dimensional hexagonal monolayer  $^{64}$  with sp<sup>2</sup> hybridized covalent bonds.<sup>21</sup> The van der Waals  $^{65}$  (vdW) interaction terms between the graphene planes provide  $^{66}$  the stability to flattened CNTs that possess reactive edges, and  $^{67}$  the curvature of which can be used to control not just chemical  $^{68}$  and mechanical properties, but thermal and electronic  $^{69}$  properties as well.<sup>22</sup>

Among polymers, epoxies containing epoxide groups and 70 71 bis-maleimides (BMI) containing maleimide groups (Figure 72 S1) are the most commonly used thermosetting resins<sup>23</sup> for making high-performance composites because they provide a 73 74 unique balance of thermal and mechanical properties 75 combined with extreme processing versatility.<sup>24,25</sup> It is worth 76 mentioning that diglycidyl ether of bisphenol A (DGEBA) and 77 4'-bismaleimidodiphenylmethane (BMPM) are some of the 78 most widely used high-performance epoxies and BMI resins in 79 the aerospace industry. For example, the fifth-generation 80 combat aircraft F-35 Lightning II developed by Lockheed 81 Martin is stated to consist of about 35% carbon fiber 82 composites out of the entire structural weight, and the primary 83 structural composites used in the aircraft are composed of 84 either epoxy or BMI matrices. Although other matrices like 85 polyethylene and polystyrene facilitate a mechanical response 86 at room temperature that satisfies property requirements for 87 some applications, there exist challenges in using such matrices 88 in more rigorous applications as it does not perform well at 89 elevated temperatures.<sup>26,27</sup> This is not the case with either 90 epoxy (DGEBA) or BMI (BMPM) as both are stable at 91 elevated temperatures.<sup>28,29</sup> Note that DGEBA and BMPM 92 monomers are often crosslinked with diethyl toluene diamine 93 (DETDA) and o,o'-diallyl bisphenol A (DABPA) monomers, 94 respectively. DETDA is an aromatic amine curing agent, and 95 DABPA is allylphenol that offers excellent toughness to the 96 BMPM.<sup>30,31</sup>

In general, a detailed understanding of the interaction 97 between graphene and a resin (hardener) matrix at the 98 molecular level is crucial for improving the composite's 99 mechanical response. This is because the adhesion between a 100 resin (hardener) and graphene is related to the chemistry of 101 the interface<sup>2,32</sup> which has, without any doubt, a considerable 102 influence on the fracture/failure behavior of a composite, 103 although it is not yet understood completely. We are aware of 104 only a few theoretical studies based on atomistic modeling that 105 were used to characterize the interfacial bonding in 106 composites.<sup>33–35</sup> Overall, it has been recognized that the 107 interface plays a significant role in the stress transfer and the 108 consequent improvements in the composite stiffness and 109 strength.<sup>36,37</sup>

In the following, we will present the results related to 111 graphene/resins (hardeners) focusing on the characterization 112 of the interface at the molecular level using a periodical 113 supercell model. The mechanical response in terms of the 114 separation point (or breaking point) and stiffness will be 115 determined. Within this atomic framework, one can directly 116 modify the interface, even with modifications that are 117 extremely difficult or impossible to control in the experiment, 118 to predict the consequences on the interfacial strength and 119 thereby its mechanical response. 120

The resin (hardener) monomers, DGEBA (DETDA) and 121 BMPM (DABPA), interacting with the surface of graphene are 122 explored. Figure 1a shows the top and side views of a 123 f1 monomer forming a complex with graphene in a periodic 124 supercell. It should be noted that our focus is not on the cured 125 polymer composites, rather on individual monomers to 126 provide a fundamental understanding of the structure– 127 property relationship in graphene–polymer composites. More-128 over, we will also consider the cases of these monomers 129 interacting with a boron nitride (BN) monolayer to benchmark 130 131 the results obtained for monomer/graphene complexes. Both
132 graphene and BN monolayer are structurally similar, but
133 electronically dissimilar materials; the covalently linked
134 graphene is a zero-gap material, whereas the semi-ionic h-BN
135 monolayer, which has a nearly identical honeycomb config136 uration as that of graphene, is semiconducting.<sup>38</sup>

## 2. COMPUTATIONAL METHOD

137 Electronic structure calculations were performed using 138 projector-augmented wave (PAW) potentials within the 139 framework of the density functional theory (DFT), as 140 implemented in the Vienna ab initio simulation package 141 (VASP).<sup>39</sup> The exchange and correlation potentials were 142 treated using a generalized gradient approximation (GGA) 143 with the Perdew–Burke–Ernzerhof (PBE) functional form.<sup>40</sup> 144 Contributions from the van der Waals (vdW) interactions, 145 which are important for graphitic systems, were incorporated 146 using Grimme's semiempirical dispersive D2 term.<sup>41</sup> The 147 energy cut-off and the convergence criterion of the energy were 148 500 and 10<sup>-5</sup> eV, respectively.

In a supercell, the periodic boundary conditions with a 149 150 lateral separation along the c-direction of 20 Å were employed 151 to ensure the negligible interaction between the periodic 152 images. The relaxation process for the equilibrium config-153 uration was carried out by placing a monomer on the top of 154 graphene in various orientations, and the distances between 155 them were set to a value that was slightly larger (or similar to) 156 than the sum of the nearest resin (hardener) atom and 157 covalent radii of carbon atoms. The structure was then relaxed 158 using the conjugate gradient algorithm until the maximum 159 forces acting on the atoms became smaller than 0.001 eV/Å. The mechanical response of the complex consisting of a 160 161 resin (or hardener) and graphene (or BN monolayer) was 162 investigated in a way that mimics the "pull-apart" experimental 163 setup commonly used for polymer composites.<sup>9,42-44</sup> In our 164 setup, a resin, which was in the equilibrium configuration, was 165 pulled up along the direction perpendicular to the surface of 166 the constrained (frozen) graphene. DFT calculations were 167 performed for each step of the pull-apart process to generate 168 the energy surface describing the interaction of a resin 169 (hardener) with graphene.

In the pull-apart setup, we define the tensile strain to be the 171 perpendicular displacement between a resin (hardener) and 172 the surface of the monolayer with respect to the equilibrium 173 separation. In this way, the strain  $\varepsilon$  can then be defined as  $\varepsilon =$ 174  $(l - l_0) l_0$ , where *l* is the displacement of the resin (hardener) 175 with respect to the distance  $l_0$  calculated in the equilibrium 176 configuration of the complex. The mechanical properties can 177 then be calculated from the strain–energy relationship,  $E_s =$ 178  $E(\varepsilon) - E(0)$ ; the energy difference between systems under a 179 given strain  $\varepsilon$  and no strain is associated with the equilibrium 180 configuration.

The calculated strain—energy curve is used to obtain the force vs strain curve, which can be used in the spinodal equation of state<sup>45</sup> to determine the critical force and critical strain of a given complex. It is worth mentioning that the spinodal equation of state, in general, applies to the bulk material, but its use has been extended to low-dimensional materials, including graphene and the BN monolayer.

188 It has been shown that the one-dimensional (1D) spinodal 189 equation of state can reproduce stress dependence on the 190 strain energy of a given system.<sup>45-47</sup> The equation of state can 191 be written as

$$\sigma = \sigma_{\rm sp} \left( 1 - \left( \frac{\epsilon_{\rm sp} - \epsilon}{\epsilon_{\rm sp}} \right)^{1/1 - \gamma} \right) \tag{1}_{192}$$

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where  $\epsilon$  and  $\epsilon_{sp}$  represent the strain at a particular point and 193 the spinodal strain, respectively.  $\gamma$  is a pseudocritical exponent 194 that varies with the direction of (stretching or compressing) 195 strain. Equation 1 can be written in terms of the force by 196 multiplying the stress terms with the effective area as follows 197

$$f = f_{\rm sp} \left( 1 - \left( \frac{\epsilon_{\rm sp} - \epsilon}{\epsilon_{\rm sp}} \right)^{1/1-\gamma} \right)$$
(2) 198

An interesting feature of the proposed force-strain equation 199 of state is that it can also be expressed analytically in its energy 200 form. Considering the isotherm at 0 K, the force is related to 201 the internal energy E and the equilibrium length L employing 202 the relationship of  $f = \frac{1}{L} \frac{dE}{d\epsilon}$ . Here,  $f_{sp}$  accounts for the 203 maximum force at which the system breaks and therefore 204 represents the elastic limit of the material. It is referred to as 205 the critical force. The value of the strain corresponding to the 206 critical force is called the critical strain. In the following 207 discussion, we designate the critical strength to represent (out- 208 of-plane) the transverse strength and the critical strain to be 209 (out-of-plane) the separation point at which the monomer 210 breaks from the monolayer. E can be understood as the energy 211 needed for the out-of-plane pull-apart (or separation) of the 212 resin (hardener) from the surface and, therefore, to overcome 213 the interatomic forces binding the complex. 214

## 3. RESULTS AND DISCUSSION

**3.1. Interaction Energy.** Initial DFT calculations were 215 performed to determine the equilibrium configurations of resin 216 (DGEBA and BMPM) and hardener (DETDA and DABPA) 217 molecules. Figure S2 displays the structural configurations of 218 these molecules, and Table S1 lists the calculated C–C and B– 219 N bond lengths ( $\sim$ 1.42 Å), which were in excellent agreement 220 with previous reports. <sup>38,48,49</sup> 221

In calculations of the energy surface describing the 222 interaction of a resin (hardener) with the monolayer, we 223 allowed it to approach the surface in the direction 224 perpendicular to the surface with a specific orientation while 225 keeping the monolayer fixed. For example, DGEBA is oriented 226 in such a way that (i) both end groups and hydrogen ends are 227 oriented toward the surface—the in-plane orientation (Figure 228 1b(i), (ii) both the epoxide end groups are facing the surface, 229 and the hydrogen end is away from the surface-the flip-in 230 orientation (Figure 1b(ii)), (iii) both the end epoxide groups 231 are aligned away from the surface, and the hydrogen end is 232 close to the surface-the flip-up orientation (Figure 1b(iii)), 233 and (iv) only one end group is facing the surface-the vertical 234 orientation (Figure 1b(iv)). At the minimum energy distance 235 associated with a specific orientation, full optimization of the 236 conjugated system in which all atoms were free to relax was 237 performed. 238

The characteristics of the complexes are determined in terms 239 of the interaction energy, the area of contact, the population of 240 atoms, Bader's charge, <sup>50</sup> critical force, and critical strain. We 241 define the interaction energy ( $\Delta E$ ) as E (complex) – E (resin 242 (hardener)) – E (monolayer), and a negative value of  $\Delta E$  243 suggests the stability of the complex.  $\Delta E$  can also be 244

245 considered as the adsorption energy of a resin (hardener) on 246 the monolayer. The contact area is the area of the resin 247 (hardener) projected on the surface of the monolayer. The 248 population of atoms is defined as the number of the resin 249 (hardener) atoms within the range 3 Å from the surface. Note 250 that the interplanar distance of  $\sim$ 3 Å can be taken as the 251 characteristic distance of vdW-bound systems including the 252 complexes considered.<sup>51</sup>

Figure 1c shows the calculated values of the interaction the energies of DGEBA and BMPM forming a complex with the presence of the results show the in-plane orientation to be the energetically preferred, and it is followed by flip-in, flip-up, and the vertical orientations. The interaction energy steadily increases the vertical orientations. The interaction energy steadily increases the vertical orientations is the interaction energy steadily increases the vertical to in-plane orientations, the vertical to in-plane orientations, the vertical to be aligned the surface. Table S2 lists the interaction energy, the the vertical of atoms of the equilibrium configurations of the the vertical cases.

Next, we calculate the effective area of contact using the 2.64 265 estimated length and width of the surface covered by a resin 266 (hardener) using its distance of one end to the other in both x-267 and y-directions. We find that the calculated area for the "in-268 plane" orientation is significantly higher than those of the rest 269 of the orientations for both cases (Table S2). This is further 270 corroborated by calculations of the population of atoms by 271 counting the number of atoms that are within the range of  $\sim 3$ 272 Å above the surface. The results show that the in-plane population is significantly larger than those of the other three 273 274 orientations, thereby facilitating a larger degree of interaction 275 between the resin atoms and graphene at the interface. 276 Subsequently, the in-plane orientation leads to the energeti-277 cally preferred configuration for DGEBA and BMPM on graphene. We note that our results are in line with the PBE 278 279 (DFT) results on the DGEBA/CNT complex, reporting the 280 preference of the longitudinal orientation over the transverse 281 orientation of DGEBA on a CNT.<sup>52</sup>

Both resins and hardeners consist of aromatic rings and side 282 groups containing oxygen or nitrogen atoms. However, 283 284 graphene consists of a honeycomb lattice, in which C atoms 285 form in-plane sp<sup>2</sup> hybridized bonds together with out-of-plane 286  $\pi$ -bonds. The interaction of a resin (or hardener) with 287 graphene is, therefore, expected to be governed by  $\pi - \pi$ 288 noncovalent interactions with a small contribution from the 289 polarizable O and N atoms of the side groups. This is what has 290 been affirmed by the analysis of the interface in terms of the area of contact and the population of atoms; a large effective 291 contact area and population of atoms yield the in-plane 292 orientation to be the energetically preferred configuration of a 293 294 resin (hardener) on graphene.

Figure S3 displays the (in-plane) equilibrium configurations period of the complexes, and the calculated values of the corresponding interaction energy values are listed in Table 1. Protection strength is DGEBA < BMPM for resins and DETDA < DABPA for hardeners. Specifically, the difference the interaction strengths of DGEBA and BMPM is or noticeable and can be explained in terms of the population of strength is significantly larger for DGEBA (Table 1). Furthermore, the polarizable O atoms (of DGEBA) induce a strength of states, though the H (resin)–C (graphene)

Table 1. Interaction Energy ( $\Delta E$ ), the Population of Atoms, and Bader's Charge (Q) Calculated for Resin and Hardener Complexes

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	complex (in- plane)	$\frac{\Delta E}{(eV)}$	population of atoms (%)	$\begin{array}{c} Q_{\mathrm{graphene}} \ (e) \end{array}$
resin	DGEBA/ graphene	-1.27	16	0.03
	BMPM/ graphene	-1.09	10	0.05
	DGEBA/BN	-1.33	16	0.03
	BMPM/BN	-1.30	15	0.06
hardener	DETDA/ graphene	-0.96	20	0.04
	DABPA/ graphene	-0.74	11	0.02
	DETDA/BN	-1.24	20	0.02
	DABPA/BN	-0.86	11	0.02

hybridized states dominate the valence band in both resin 307 f2 complexes (Figure 2c,d). 308 f2

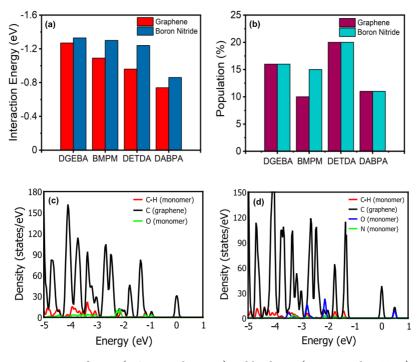
For the case of the BN monolayer, we expect a small but 309 noticeable difference in the calculated results as compared to 310 those obtained for graphene. This is because the BN 311 monolayer being semi-ionic may induce a slightly higher 312 degree of electrostatic interaction with a resin (hardener) 313 relative to graphene. The calculated values of the interaction 314 energy are higher;  $DGEBA_{(graphene)} < DGEBA_{(BN)}$  and 315  $BMPM_{(graphene)} < BMPM_{(BN)}$  for the resin complexes and 316  $DABPA_{(graphene)} < DABPA_{(BN)}$  and  $DETDA_{(graphene)} < 317$   $DETDA_{(BN)}$  for the hardener complexes. 318

Moreover, the results indicate that curing agents (e.g., 319 DETDA) participate effectively at the interface, which implies 320 that the load transfer of epoxy/graphene composites is 321 governed by both the resin and the hardener in this case. 322 The interaction energy values for DETDA and DABPA are in 323 line with the expectations arising from their values of the 324 population of atoms (Table 1). It is worth mentioning that the 325 aromatic ring of DETDA is aligned parallel to graphene and 326 thereby yielding a higher degree of the vdW interaction with 327 monolayers at the interface.

Overall, the interaction between the resin (or hardener) and 329 graphene (or BN monolayer) is governed by the dispersive 330 force like vdW in the cases considered. This is due to the fact 331 that the analysis of the calculated Bader charge suggests the 332 occurrence of a small charge transfer (<0.1 e) from graphene 333 (BN monolayer) to resins (hardeners) (Table 1). Moreover, 334 the hydrogen-bonding interactions with  $\pi$  electrons of 335 graphene (or B/N atoms of BN monolayer) at the separation 336 of 2.5 Å are expected to be weak.<sup>53</sup>

**3.2. Mechanical Response.** The (out-of-plane) mechanical response of the complexes was calculated using the setup 339 and by applying the load in only the "*z*"-direction, which is 340 perpendicular to the surface starting from its equilibrium 341 separation to the separation of  $\approx 3.8$  Å with a step-size of 0.02 342 Å (Figure S4). DFT calculations were performed at each step 343 generating the energy surface of the resin (DGEBA or BMPM) 344 and the hardener (DABPA or DETDA) forming complexes 345 with graphene (or BN monolayer) (Figure S5). The strain (or 346 separation)-energy surfaces were then used to calculate force 347 vs strain curves, as shown in Figure 3. Subsequently, the force 348 fs vs strain curve was used in the 1D spinodal equation of state 349 (eq 2). We noted that two parameters,  $f_c$  and  $\varepsilon_c$ , defined the 350 shape of each curve, which was used to characterize the 351

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**Figure 2.** (a) Calculated interaction energy of resins (DGEBA and BMPM) and hardeners (DABPA and DETDA) with graphene (red) and the BN monolayer (blue). (b) Calculated population of atoms within the range of 3 Å distance above the monolayer. Atom-projected density of states for (c) resin DGEBA with graphene and (d) resin BMPM with graphene.

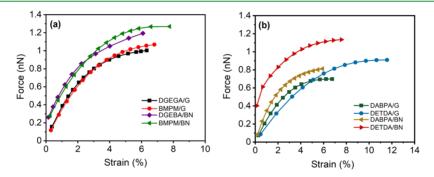
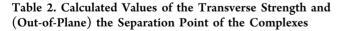


Figure 3. Calculated force vs transverse strain curves of the (a) resin (DGEBA or BMPM) and (b) hardener (DABPA or DETDA) forming complexes with graphene (or BN monolayer).

<sup>352</sup> transverse strength and the separation point of the conjugated <sup>353</sup> monomers considered (Table 2).  $\gamma$  is the 1D correspondence <sup>354</sup> of the pseudocritical exponent  $\beta$ , which, in general, takes a <sup>355</sup> value of 0.85 associated with the volumetric compression of

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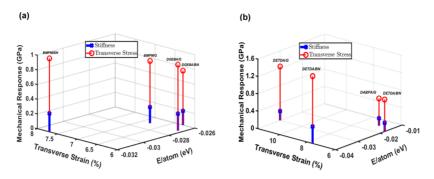


		complex	transverse strength $f_{\rm c}$ (nN)	(out-of-plane) separation point $\varepsilon_{\rm c}$ (%)
	resin	DGEBA/ graphene	1.0	6.3
		BMPM/ graphene	1.1	6.8
		DGEBA/BN	1.2	6.1
		BMPM/BN	1.3	7.7
	hardener	DETDA/ graphene	0.9	11.6
		DABPA/ graphene	0.7	6.7
		DETDA/BN	1.1	7.6
		DABPA/BN	0.8	5.9

solids.<sup>54,55</sup> In the case presented,  $\gamma$  varies between 0.5 and 0.8 356 (Table S3). 357

It is worth mentioning that the equation of state (e.g., eq 1), 358 in general, is used to investigate the constitutive responses of a 359 bulk-like system. In the case presented, we have used the 360 spinodal equation of state to differentiate the (out-of-plane) 361 mechanical response of quasi-three-dimensional (3D) com- 362 plexes consisting of a resin (hardener) and graphene (BN 363 monolayer). Therefore, in the following discussion, we use 364 "transverse" to describe the out-of-the-plane and designate the 365 critical strength to represent the transverse strength and the 366 critical strain to be (out-of-plane) the separation point at 367 which the monomer apart from the monolayer (Figure S6). 368

The calculated results show that the resin (hardener)  $_{369}$  forming complexes with the BN monolayer exhibit a slightly  $_{370}$  higher transverse strength than the corresponding complexes  $_{371}$  with graphene. Among resins, BMPM exhibits a slightly higher  $_{372}$  transverse strength relative to that of DGEBA. This is not the  $_{373}$  case with the separation point values for which we find the  $_{374}$  order to be DGEBA/graphene  $\approx$  DGEBA/BN < BMPM/  $_{375}$  graphene < BMPM/BN. Interestingly, the hardener DETDA  $_{376}$ 



**Figure 4.** Interaction energy/atom vs transverse strain and stiffness for the (a) resin (DGEBA or BMPM) and (b) hardener (DABPA or DETDA) forming complexes with graphene (or BN monolayer).

377 exhibits a relatively high value of the separation point 378 suggesting that, as a curing agent, it is likely to improve the 379 interfacial load transfer of the epoxy/graphene composite. It is 380 worth noting that the calculated transverse strength of 381 DGEBA/graphene is comparable to the calculated value of 382 DGEBA/CNT; MD calculations reported the value of the 383 force to be ~3.8 nN, which breaks a DGEBA strand from a 384 CNT,<sup>29</sup> whereas we find the value of ~1 nN to break the 385 DGEBA from graphene (Table 2).

In general, the (out-of-plane) stiffness can be taken as a measure of the quasi-3D Young's modulus suggesting that a higher value of the stiffness leads to a higher Young's modulus, which is one of the desired characteristics of an engineered composite.

Table S4 lists the values of the transverse stress and the 391 392 stiffness calculated from the stress-strain curve displayed in 393 Figure S7 for which the area is estimated by projecting the 394 length and width of the resin (hardener) over graphene (or BN monolayer). The results show that BMPM/graphene (or 395 396 BMPM/BN) is stiffer than the corresponding DGEBA complexes. Among the hardeners considered, DETDA having 397 (graphene-like) planar structure is predicted to have higher 398 а stiffness than DABPA in the complexes considered. Next, we 399 400 benchmark our stiffness values to the values calculated for 401 bilayer graphene employing a similar pull-apart setup. The 402 elastic stiffness of bilayer graphene is calculated to be 24 GPa, 403 an order of magnitude higher than what is predicted for the 404 resin (or hardener) forming complexes with graphene. This is 405 in line with the expectation of the occurrence of a relatively 406 high degree of vdW interactions at the interface with a 100% 407 population of atoms for the case of bilayer graphene.

<sup>408</sup> Finally, Figure 4 summarizes the result of the present study <sup>409</sup> displaying the predicted relationship between the interaction <sup>410</sup> strength at the interface and the mechanical response in terms <sup>411</sup> of stiffness obtained at the molecular level for epoxy and BMI <sup>412</sup> composites. The results show that the mechanical response is <sup>413</sup> directly related to the degree of the interface adhesion, though <sup>414</sup> it can also be influenced by the nature of the chemical bonds at <sup>415</sup> the interface in the polymer composites.

#### 4. SUMMARY

416 We performed an atomistic investigation of the representative 417 polymer composites employing the density functional theory 418 to establish the relationship between the nature of the interface 419 and its mechanical response. Specifically, DGEBA (DETDA) 420 and BMPM (DABPA) were considered to represent resin 421 (hardener) components of the epoxy and BMI, respectively. 422 The calculated results indicated that the interfacial adhesion

was highly influenced by the orientation of the resin 423 (hardener) on graphene due to van der Waals interactions 424 being dominated at the interface. Next, we used the strain- 425 energy relationship to extract force vs strain curves, which were 426 then used in the 1D spinodal equation of state to determine 427 the mechanical response of a complex. We found that the 428 mechanical response in terms of stiffness follows the 429 hierarchical order of the interaction strength at the interface; 430 elastic stiffness of BMPM/graphene was higher than that of 431 DGEBA/graphene. The change in polarity of the surface from 432 graphene to the BN monolayer improved the interfacial 433 strength and thereby the elastic stiffness due to the presence of 434 covalent polar bonds at the interface. Furthermore, the 435 presence of aromatic rings in the hardener DETDA yielded a 436 significantly higher mechanical response relative to the one 437 exhibited by the DABPA hardener suggesting that DETDA, as 438 a curing agent, was likely to improve the interfacial load 439 transfer of the epoxy/graphene composite. Overall, the results 440 predicted that a small degree of polarity at the interface 441 dominated by van der Waals interactions can help in improving 442 its mechanical response. In light of the intricate molecular-level 443 simulations based on first-principles methods, predicting the 444 interfacial properties of composites is challenging but can 445 provide an insight to experimentalists in tailoring composites 446 properties as demanded by aerospace applications. 447

#### ASSOCIATED CONTENT

#### **Supporting Information**

The Supporting Information is available free of charge at 450 https://pubs.acs.org/doi/10.1021/acsapm.0c01306. 451

Epoxide and maleimide groups; structural configurations 452 and geometrical parameters of resin and hardener 453 monomers; orientation-dependent properties of com- 454 plexes; pull-apart force mechanism; mechanical response 455 of the complexes simulated in a periodic supercell 456 (PDF) 457

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## Corresponding Authors Geeta Sachdeva – Department of Physics, Michigan

Technological University, Houghton, Michigan 49931,	461
United States; Email: gsachdev@mtu.edu	462
Ravindra Pandey – Department of Physics, Michigan	463
Technological University, Houghton, Michigan 49931,	464
United States; @ orcid.org/0000-0002-2126-1985;	465
Email: pandey@mtu.edu	466

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449

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459

460

#### 467 Authors

468 Alvaro Lobato – MALTA-Consolider Team and

- 469 Departamento de Química Física y Analítica, Universidad de
  470 Oviedo, E-33006 Oviedo, Spain
- 471 Gregory M. Odegard Department of Mechanical
- 472 Engineering and Engineering Mechanics, Michigan
- 473 Technological University, Houghton, Michigan 49931,
- 474 United States; (a) orcid.org/0000-0001-7577-6565

475 Complete contact information is available at:

476 https://pubs.acs.org/10.1021/acsapm.0c01306

#### 477 Notes

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